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OGDEN AIR LOGISTICS CENTER

UNITED STARES AIR FORCE

HILL AIR FORCE BASE, UTAH 84406



SURVEILLANCE REPORT'

STAGE I

DISSECTED MOTORS

PHASE IX

PROPELLANT AND COMPONENT TESTING

Quarterly rept.

PROPELLANT LAB SECTION

MANCE- COOK A. / Thompson

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MANCP REPORT NR 392(78)
MMEMP PROJECT M82934C-WNL17514

# SURVEILLANCE QUARTERLY REPORT

STAGE I DISSECTED MOTORS

PHASE IX PROPELLANT & COMPONENT TESTING

Author

JOHN A. THOMPSON, Chemist Component & Combustion Test Unit

Engineering & Statistical Review By

JOHN K. SCAMBIA, Project Engineer
Service Engineering

EDWARD J. ZRICKSON, Statistician

Data Analysis Unit

Recommended Approval By

Konald H. Lansen RONALD F. LARSEN, Chief

Physical & Mechanical Test Unit

LEONIDAS A. BROWN, Chief

Component & Combustion Test Unit

on F Woodl

DON F. WOODS, Chief Propellant Laboratory Section

March 1978

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Industrial Products & Ldg Gear Division
Directorate of Maintenance
Ogden Air Logistics Center
United States Air Force
Hill Air Force Base, Utah 84406

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#### ABSTRACT

Testing was performed to determine the useful shelf/service life for LGM-30 Stage I Rocket Motors. A three year storage program for propellant and components was started in May 1961. This program was then extended to a ten year study and later continued indefinitely to assure that a deterioration in motor physical characteristics could be detected in time to take some corrective actions before the weapon system performance deteriorated below an acceptable level.

This report covers only propellant data and limited case bond data. The malfunction of an environmental chamber destroyed component samples that had originally been part of this testing program (and the inadvertent burning of some motors during dissection reduced the material available for testing). Planned dissection of selected motors in the future will provide samples for continued component testing. Test specimens for this reporting period were obtained from motors STM-012, 0012099, 0012199 and UP7775 block propellant.

Separate analyses were made on the respective motors and block propellant for the first time in this report and are shown in the regressions. The plotting symbols for each motor and block propellant are listed in the statistical analyses section.

The data from this test period was combined with data from previous testing and entered into the GO85 computer for storage, analysis, and regression analysis. From the statistical analysis of all data tested to date, significant degradation of the propellant does not appear likely for at least two years past the oldest data point.

Future testing will be conducted on dissected motors.

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#### GLOSSARY OF TERMS AND ABBREVIATIONS

Aging Trend A change in properties or performance result-

ing from aging of material or component

CSA Cross Sectional Area

DB Dogbone

Degradation Gradual deterioration of properties or performance

E Modulus (psi), defined as stress divided by strain along the initial linear portion of the

curve.

EB End Bonded

EGL Effective Gage Length

em Strain at maximum stress

er Strain at rupture

"F" ratio The ratio of the variance accounted for by the

regression function to the random unexplained variance. The regression function having the most significant "F" ratio is used for plotting data. The ratio is also used in detecting signi-

ficant changes in random variation between

succeeding time points

JANNAF Joint Army, Navy, NASA, Air Force Committee

MANCP Propellant Lab Section at Ogden Air Logistics Center

Ogden ALC Ogden Air Logistics Center, Air Force Logistics

Command

r or R The Correlation Coefficient is a measure of the degree

of closeness of the linear relationship between two

variables

Regression The general form of the regression equation

Equation is Y = a + bx

Regression Line representing mean test values with respect

Line to tim

S<sub>b</sub> Standard error of estimate of the regression

coefficient

# GLOSSARY OF TERMS AND ABBREVIATIONS (cont)

Se or Sy.X Standard deviation of the data about the

regression line

Sm Maximum Stress

Sr Stress at rupture

Standard Square root of variance

Deviation  $(S_y)$ 

Strain Rate Crosshead speed divided by the EGL

"t" test

A statistical test used to detect significant differences between a measured parameter and an expected value of the parameter (determines if

regression slope differs from zero at the 95%

confidence level)

Variance The sum of squares of deviations of the test

results from the mean of the series after division by one less than the total number of test

results

3 Sigma Band The area between the upper and lower 3 sigma

limit. It can be expected that 99.73% of the inventory represented by the test samples would fall within this range assuming that the popu-

lation is normally distributed.

90-90 Band It can be stated with 90% confidence that 90% of

the inventory represented by the test samples would fall within this range assuming that the

population is normally distributed

# INTRODUCTION

#### A. PURPOSE:

This report contains test data from samples of LGM-30 Stage I, Wings I-V TP-H1011 propellant and case bond specimens. Testing was performed by the Propellant Laboratory Section (MANCP) for the Engineering and Reliability Branch of the Airmunitions Management Division (MMWRM) under Project M82934C-WNL17514. This report is the ninth in this series. Data from this test period and propellant test data from the eight previous reports were entered into the GO85 computer for regression analysis. The regressions are shown in this report.

# B. "EST PROGRAM:

The LGM-30 Laboratory and Component Program includes the testing of materials used in the main case, aft closure, ignition assemblies, and main grain propellant. This report covers TP-H1011 propellant and case bond specimens. Table I outlines the test program.

Propellant for testing was obtained from three dissected motors, STM-012, a motor prepared by Thiokol specifically for dissection, S/N 0012099, A SLIM motor, and motor S/N 0012199 selected from the inventory for dissection and UP-7775 block propellant.

# C. HISTORICAL BACKGROUND:

In May 1961, Thiokol began a three year LGM-30 laboratory storage and test program to determine the rate of degradation with age for Stage I materials. During June 1962 and again in August 1963, additional samples were included. New samples were added in July and August 1964 when the

surveillance testing program was extended to ten years (Test Plan 0717-62-0967, 53-8). Carton block propellant, batch UP-7775, containing TP-H1011 propellant cast in March 1962 was added to the program in 1964.

Samples added to the inventory in 1964 were considered to be a new population, but were combined in regression analysis with the three dissected motors. The history of testing of these materials is found in MQQP Report Nrs. 109A(67), 144(68), 208(71) and 358(76). Physical transfer of the specimens from Thiokol to Ogden ALC was made in June 1967.

TABLE I

# TEST PROGRAM

# All Temperatures in Fahrenheit

STM-012, SN 0012099, SN 0012199, UP-77.75

Test	Conditions	Spec/ Cond	Spec Conf
Tensile	77°, 2.0 & 20 in/min	5 ea	JANNAF Dogbone
Creep	77°, 10 & 12 lb load	3 ea	JANNAF Dogbone
Stress Relax	77°, 3 & 5% strain	3 ea	1/2" x 1/2" x 4"
Strain Dilatation	77°, .25 in/ in/min	3 еа	1/2" x 1/2" x 4"
Hardness	77°, initial 7 10 sec	5 ea .	Dogbone Ends
ное	77°	3 ea	1/2" x 3/8" x 1"
Burning Rate	77", 500 & 1,000 psi	5 ea	.156" x .156" x 5"
DTA	77° start	3 ea	0.040" wafer
Ignitability	77°, 168 cal/ cm <sup>2</sup> /sec	3 ea	0.050" wafer
Sol Gel	77°	6 ea	1/2" x 1/2" x 1/2"
UP 7775 will <u>NOT</u> be	e used on the following te	sts:	
High rate Tensile	77°, 1,000 in/in/min	5 ea	3/4" GL Dogbone
Triaxial High Rate	77°, 1,000 in/in/min 600 psi	3 ea	3/4" GL Rail
Dynamic Response	77°, 70 gm ct wt	3 ea	3.3" x .33" x 0.690" disc
Biaxial constant Strain	77°	3 ea	3/4" GL Rail

TABLE I (cont)

	TABLE I (con	nt)	
Test	Conditions	Spec/Cond	Spec Conf
Failure Envelope	Temp: -50° -20°, 10° 40°, 77° 130°, & 180°F at a rate of 0.2, 2.0, & 20 in/min	3 ea	JANNAF Dogbone
STM012 and 001219	9 only will be for the	following tests:	
Case Bond Tensile	77°, 0.2 in/	10 ea	1" x 5/8" x 3/4"
Tear Energy	77°F <u>+</u> 2°	8ea	0.1" x 1.18" x 3"
Poisson's Ratio (Strain Dilatation) 10, 15, 20 25, 30%	77°F ± 2°	6 per/condition	0.50" x 0.50' x 4"
	- 4	-	

#### STATISTICAL ANALYSIS

The objective of this statistical analysis is to determine whether or not any aging trends are demonstrated by accumulated test data in order to assist Service Engineering to more accurately predict motor serviceability.

Propellant was made available for testing and statistical analysis to obtain an overall view of the aging trends affecting the First Stage Dissected Motor Program. The sampling consists of data from two dissected operational motors (0012099 and 0012199), and carton propellant (batch UP-7775). One motor (STM-012) was prepared by Thiokol specifically for the dissection program.

A Multi-symbol Regression Analysis Program was used to determine aging trends. The sampling is combined for each test parameter in a single regression analysis. The linear equation (Y = a + bX) was found to be the best fit model for the data in this report. A composite population aging trend line is then calculated accepting the fact that individual aging of different populations may be masked.

The Multi-symbol Program uses a unique plotting code for each motor and carton data on the regression plots. This method of data plotting allows a visual display of the overall relationship between the various origins of data and how they relate to the overall least square aging trend line.

The regression program uses an analysis with individual data points from different time periods combined to establish a least squares aging trend line for the overall data. The variance about the regression line, obtained using individual values of the dependent variable, was used to compute a tolerance interval such that at the 90% confidence level 90% of the population falls within this interval. This tolerance interval was

extrapolated to a maximum of 24 months to give an indication of the statistical significance of the slope of any aging trends. The computer tolerance interval about the composite regression line is wider than what the tolerance interval would be about any individual motor or carton regression line because of the increased data spread introduced by combining different populations of data. The "t" values and the significance of this statistic, which are reported for each regression model, gives an indication of the "statistical significance" of the slope of the aging trend in the Y-axis. Data and regression trend lines were plotted utilizing an IBM-360/65 computer.

# ORIGIN SYMBOL TABLE

Origin	DOM	Wing	Symbol Symbol
Motor 0012099	63166	2	0
Motor 0012199	63227	2	1
Motor STM-012	61221	1	S
Carton UP-7775	62075	1	U

#### TEST RESULTS

Regression analysis is the method of evaluation used in the analysis of the test results.

### A. TENSILE:

Low rate (2.0 in/min) tensile testing data shows a statistically significant gradual decrease for strain at maximum stress and strain at rupture. Modulus shows a statistically significant gradual increase. Maximum stress and stress at rupture shows no significant change (Figures 1 thru 5).

The 20 in/min low rate tensile testing shows a statistically significant gradual decrease for strain at maximum stress and stress at rupture with the other regressions showing no significant change (Figures 6 thru 10).

The high rate strain parameters show a statistically significant gradual decrease (Figures 11 and 13). The stresses and modulus do not show a statistically significant change (Figures 12, 14 and 15).

High rate triaxial testing shows a statistically significant increase for strains and stresses (Figures 16 thru 19). The modulus shows a statistically significant decrease (Figure 20).

Case bond tensile shows a statistically significant decrease (Figure 21).

# B. CREEP:

For both the 10 and 12 pound load no significant change is shown except for the 10 pound load at 20 sec which shows a statistically significant gradual decrease (Figures 22 thru 29).

# C. STRESS RELAXATION:

Stress relaxation modulus for both 3 and 5% strain shows no significant change (Figures 30 thru 37).

# D. CONSTANT STRAIN:

A statistically significant decrease is shown for constant strain (Figure 38).

# E. SHORE HARDNESS:

The Shore A ten second hardness shows no significant change (Figure 39).

# F. BURNING RATE:

A statistically significant decrease is shown for both the 500 and 1,000 psi initial testing (Figures 40 and 41).

# G. HEAT OF EXPLOSION:

The heat of explosion does not show a significant change (Figure 42).

### H. IGNITABILITY:

No significant change is seen in the data by the regression (Figure 43).

# I. DIFFERENTIAL THERMAL ANALYSIS (DTA):

The endotherm and first exotherm do not show a significant change (Figures 43 and 45). The ignition temperature shows a statistically significant increase (Figure 46).

# J. SOL GEL:

A statistically significant gradual decrease is shown for percent extractables, gel swell ratio, sol gel density and cross link density (Figures 47 thru 50).

# K. FAILURE ENVELOPE:

Failure envelopes for STM-012 and 0012199 test data are shown in Figures 51 and 52 respectively.

# CONCLUSIONS

The test results show that, under present storage conditions, some of the physical/mechanical and combustion properties of the propellant indicate statistically significant aging trends. On some regressions where a significant trend is indicated, the slope of the trend line is quite gradual and no operational problems are expected. On other regressions, i.e., triaxial tensile and burning rate, the slope of the trend line appears quite steep although, in reality, the percent change is minor as indicated by the formulas found at the top of each figure. The Y-axis range is automatically varied by the data spread to provide visibility between individual data means. As a result, the range values (on Y-axis) must be considered when visually analyzing regression slopes.

Although some aging trends have been observed, it does not appear that any significant degredation will occur in the propellant within the next two years.

# RECOMMENDATIONS

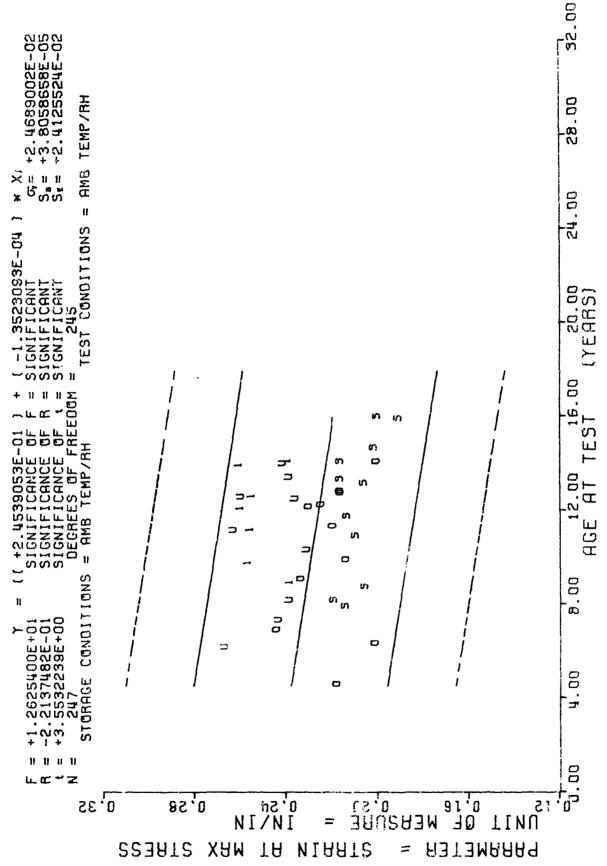
It is recommended that continued testing be conducted on the three dissected motors and UP-7775 block propellant presently being tested and also on those motors selected for testing by Service Engineering. On those motors selected for future dissection, testing will include propellant, casebond and component materials.

\*\*\* SAMPLE SIZE SUMMARY \*\*\*

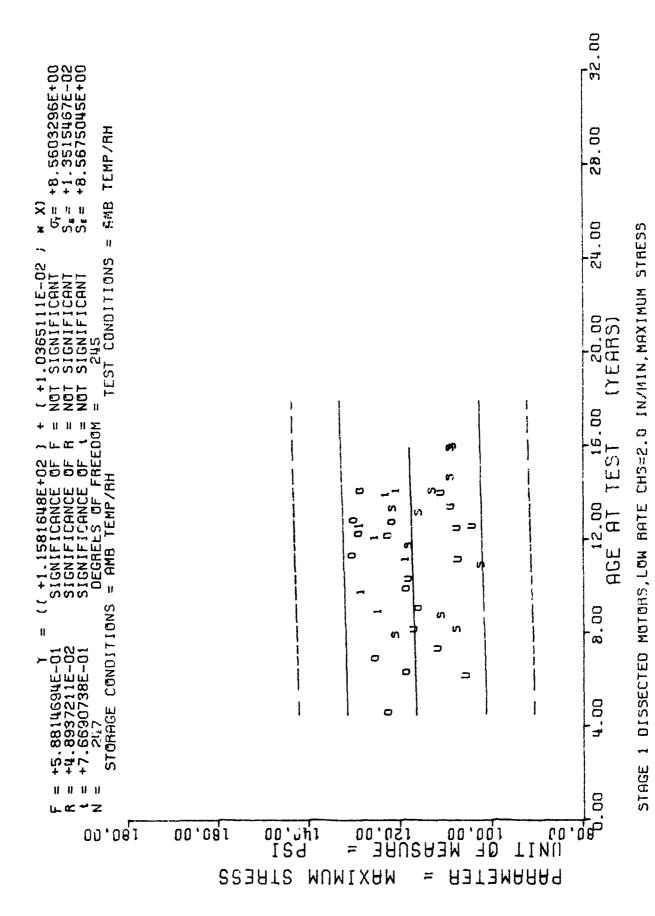
NR SAMPLES	20 M M M M M M M M M M M M M M M M M M M	
ACE (MONTHS)	159.0 166.0 166.0 167.0 168.0 175.0 191.0	
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AGE (MONTHS)		157.0

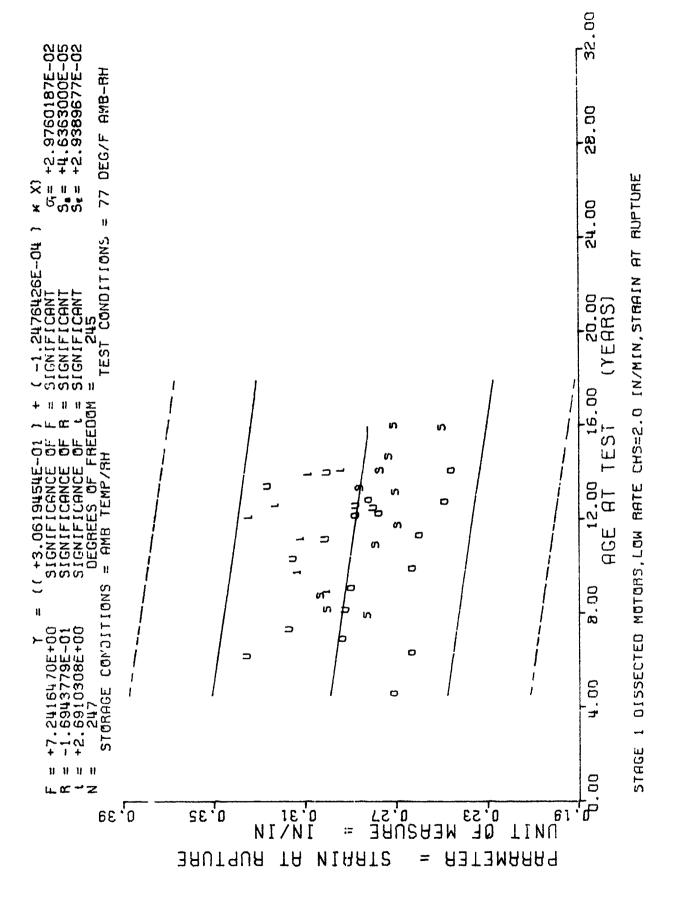
STAGE I DISSECTED MOTORS, LOW RATE CHS=2.0 IN/MIN, STRAIN MAX STRESS

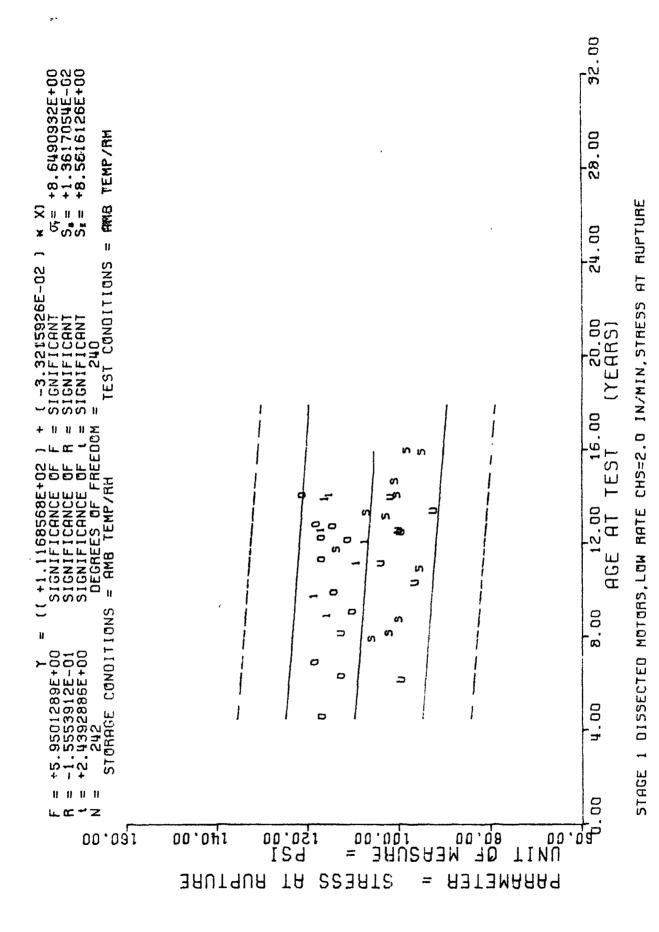
This sample size summary is applicable for figures 1 thru 4.



STRESS XAM CHS=2.0 IN/MIN, STRRIN DISSECTED MOTORS, LOW RATE STAGE 1



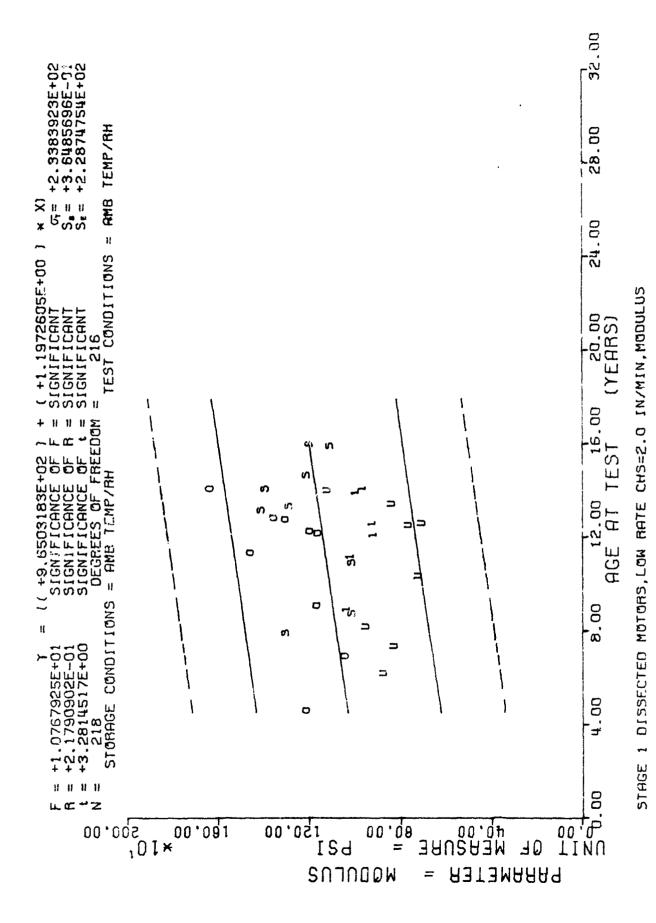




\*\*\* SAMPLE SIZE SUMMARY \*\*\*

NR SAMPLES	2 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	
AGE (MONTHS)	163.0 175.0 190.0 191.0	
SAMPLES	017 645010 950 950 950 950 950 950 950 950 950 95	v
AGE (MUNTHS)	4	167.0

STAGE 1 DISSECTED MUTORS, LOW RATE CHS=2.0 IN/MIN, MODULUS This sample size summary is applicable to figure 5

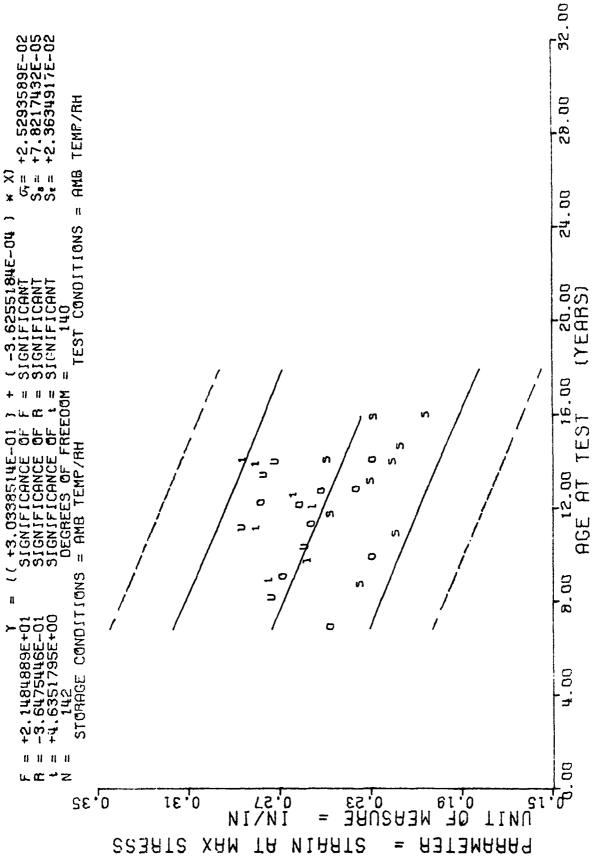


\*\*\* SAMPLE SIZE SUMMARY \*\*\*

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STARK I DISSECTED MOTERS, LOW RATE CHS=20,0 IN/MIN, STRAIN MAX STRESS

This sample size summary is applicable to figures 6 thru 9



STRESS CHS=20.0 IN/MIN, STRRIN MAX 1 DISSECTED MOTORS, LOW RATE STAGE

9

Figure

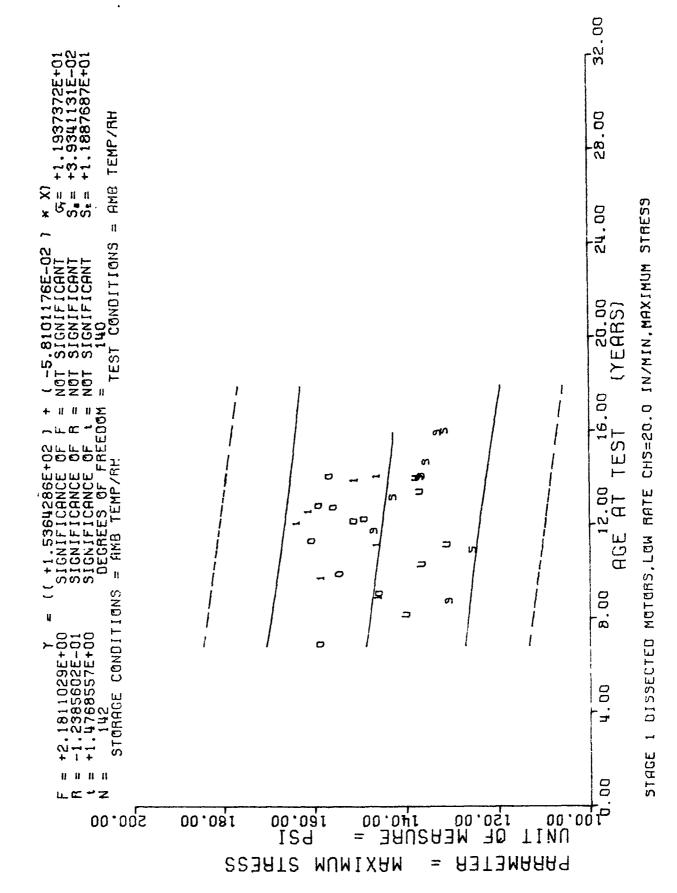


Figure 7

Figure 8

RUPTURE

H

CHS=20.0 IN/MIN, STRRIN

DISSECTED MOTORS, LOW RATE

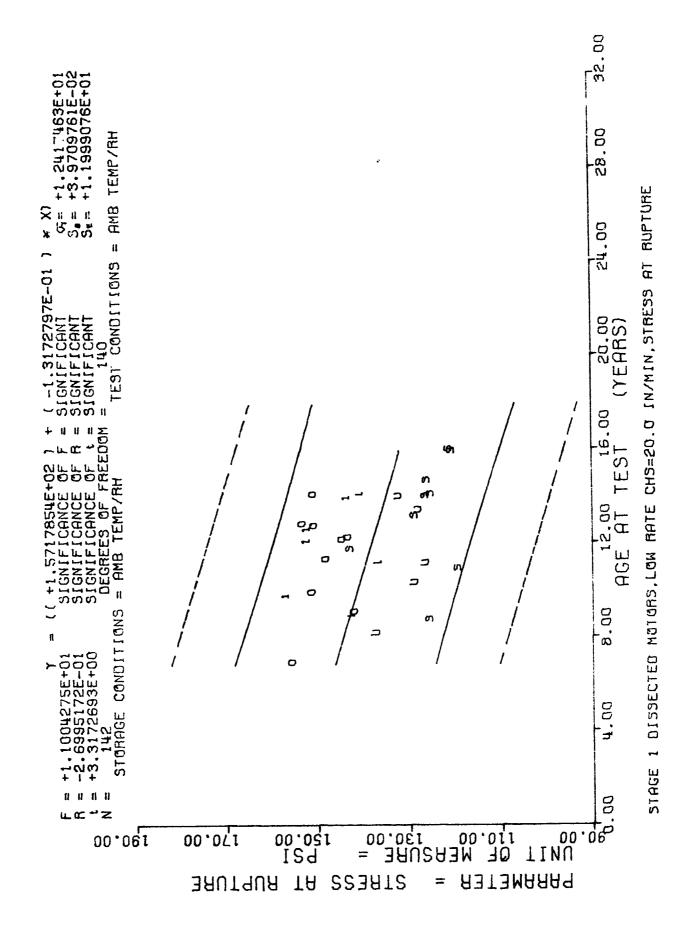
STAGE

NIAATS

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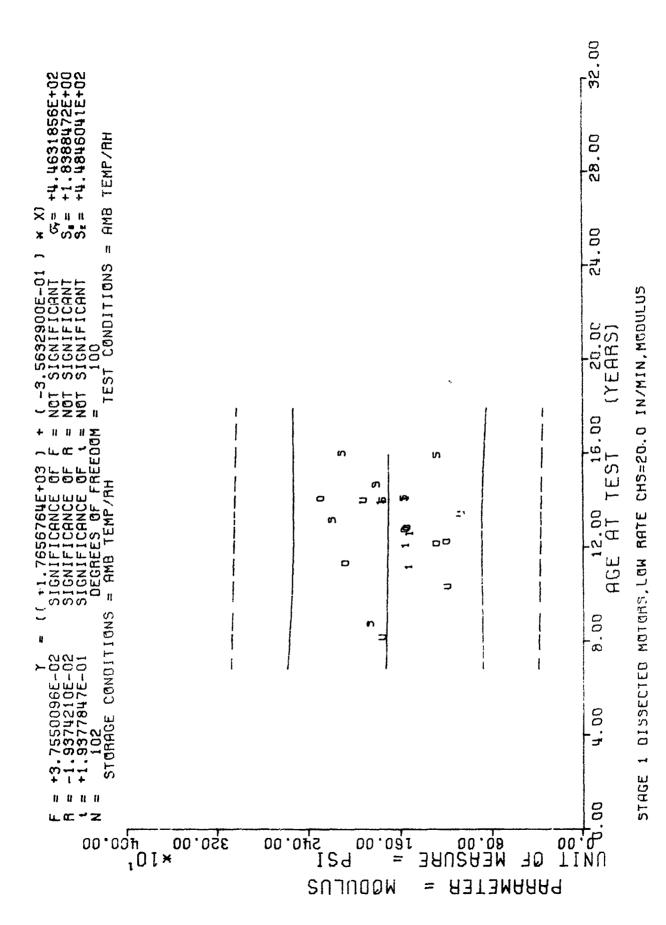
PARAMETER

BUPTURE



N3 SAMPLES	とらこと4を4mののまぐとららているまで
AGE (MJNTHS)	19 19 19 19 19 19 19 19 19 19 19 19 19 1

STAGE 1 DISSECTED MOTURS, LOW RATE CHS=20.0 IN/MIN, MODULUS This sample size summary is applicable to figure 10

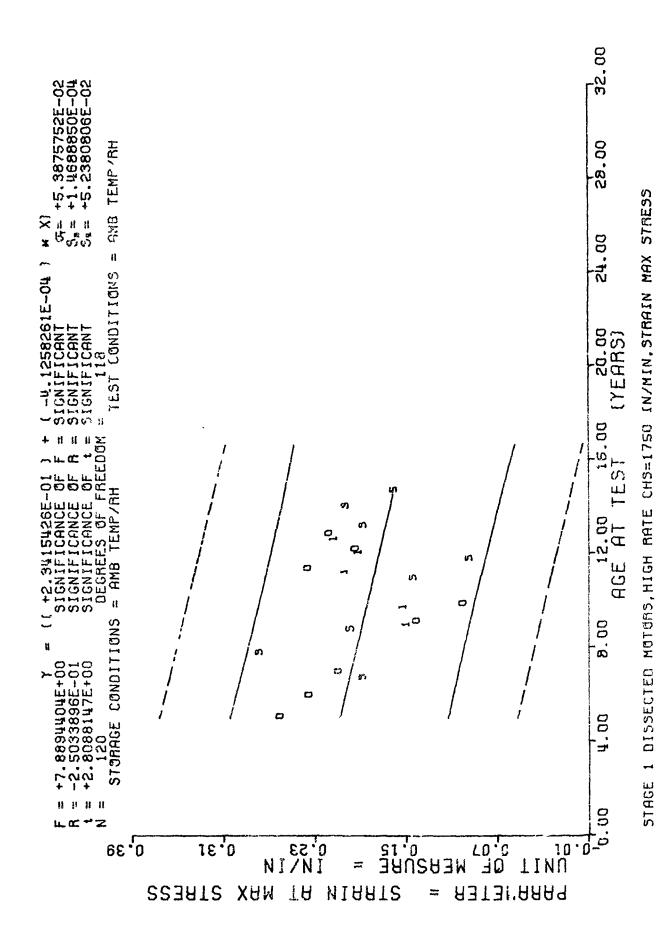


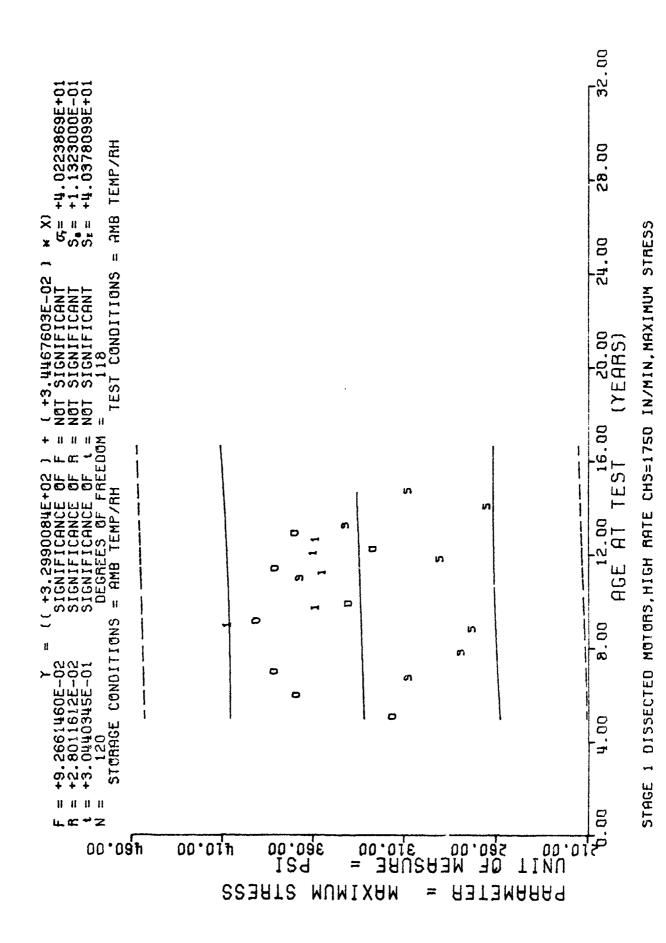
ž	SAMPLES
AGE	(MONTHS)

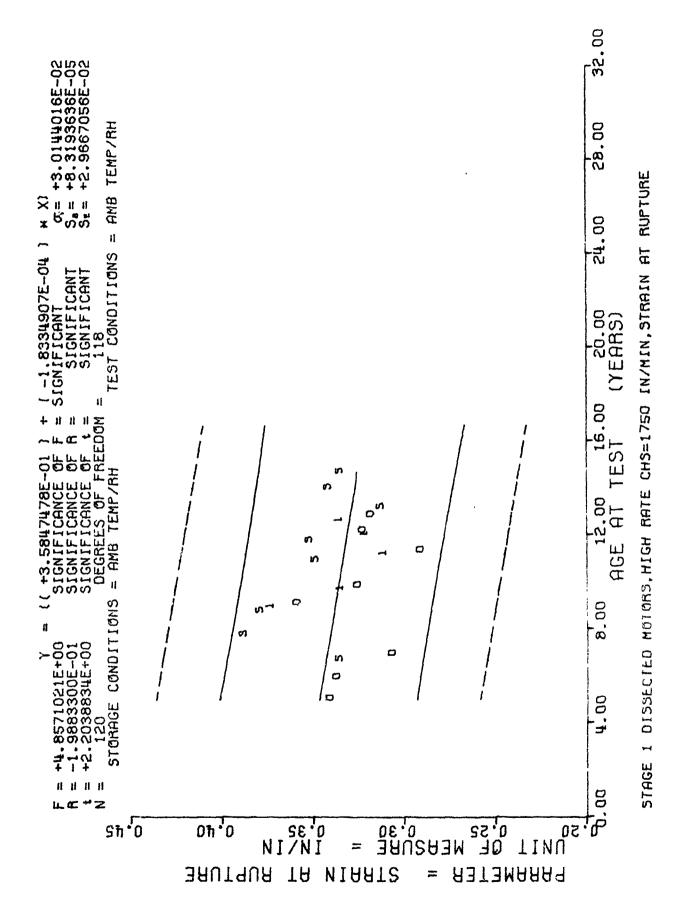
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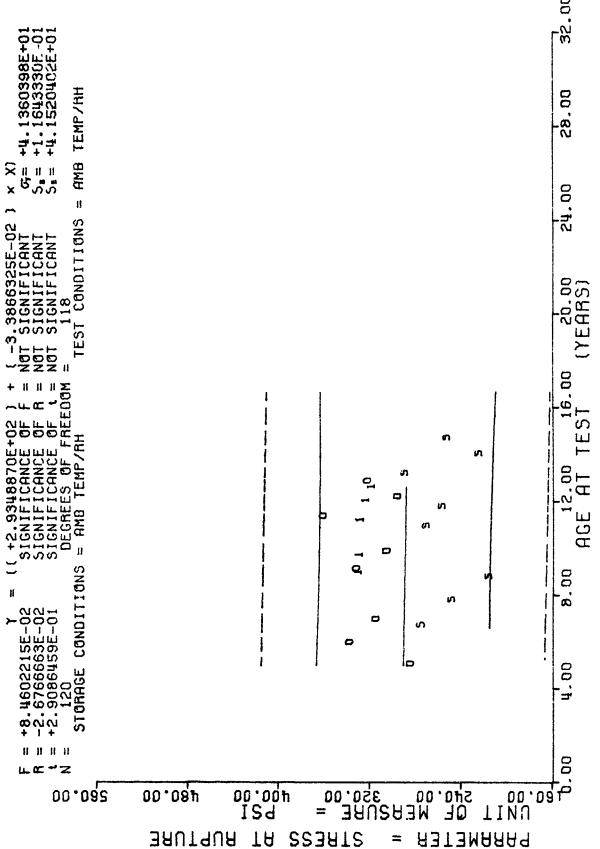
STAGE 1 DISSECTED MOTURS, HIGH RATE CHS=1750 IN/MIN, STRAIN MAX STRESS

This sample size summary is applicable to figures 11 thru 15



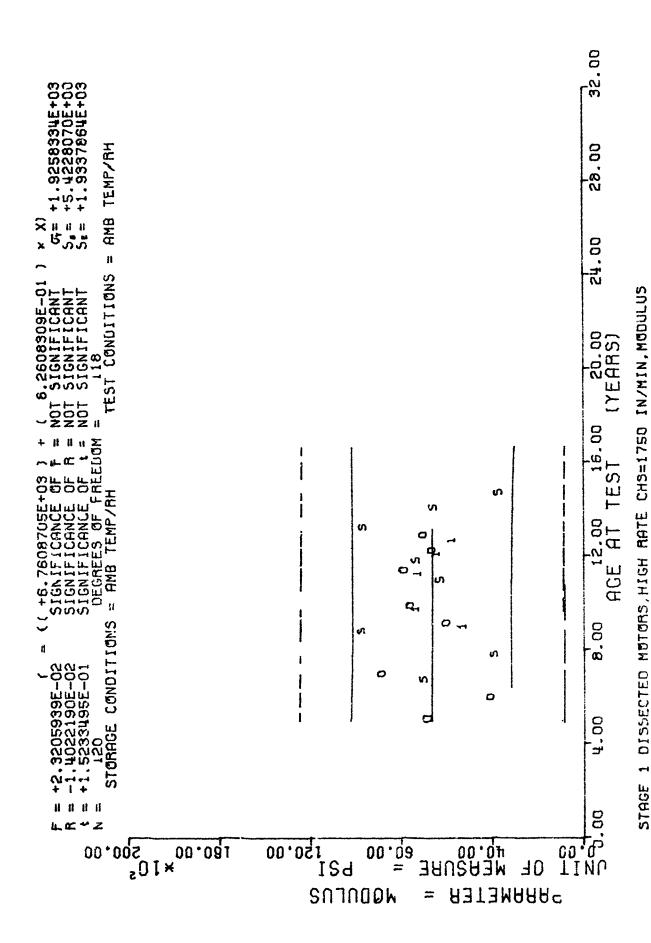






1 DISSECTED NOTORS, HIGH RATE CHS=1750 IN/MIN, STRESS AT BUPIURE STAGE

Figure 14



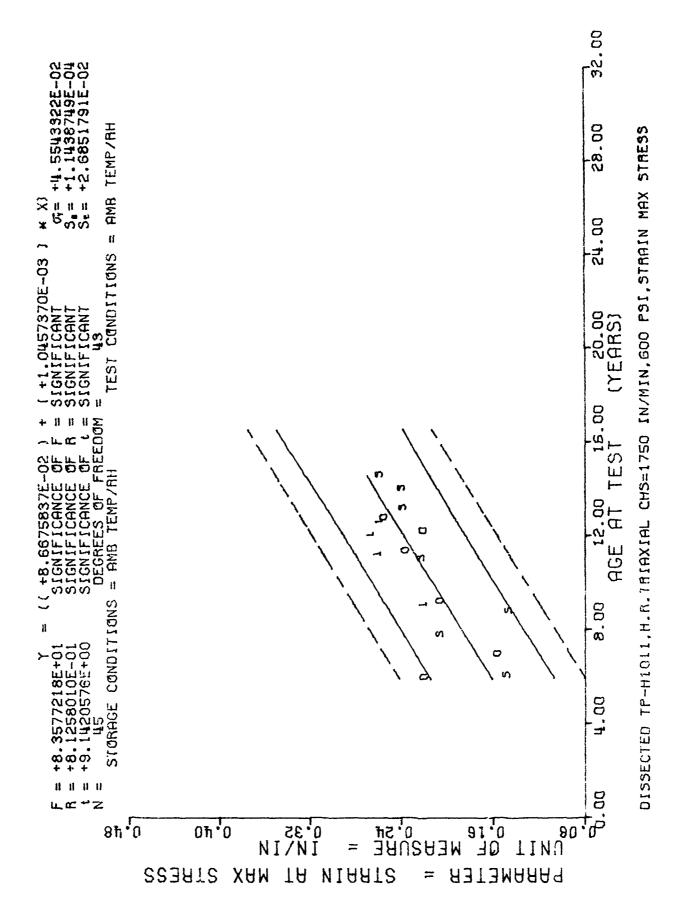
- 31 -

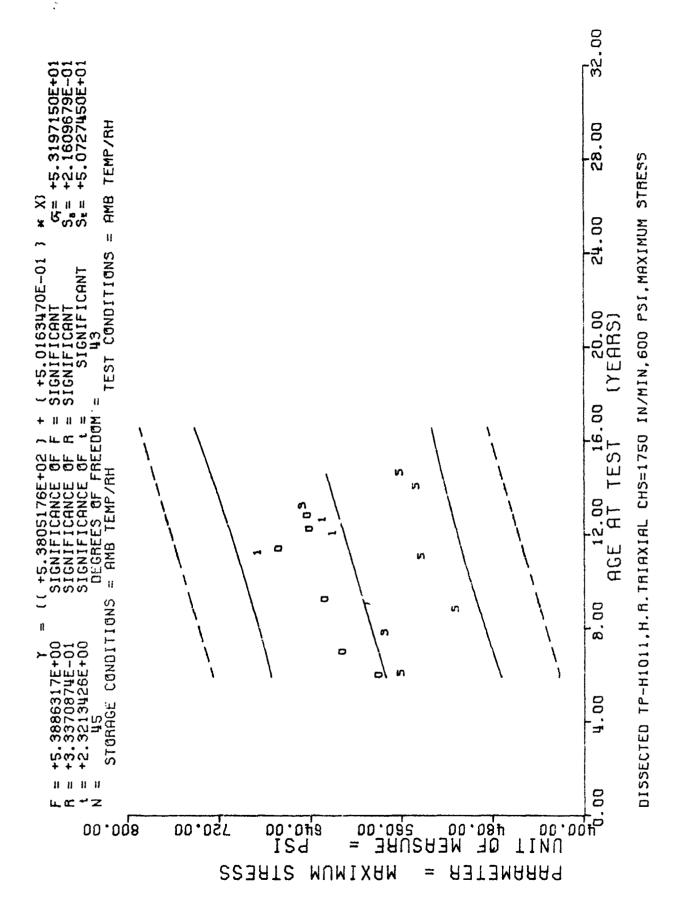
ž	SAMPLES
AGE	(MONTHS)

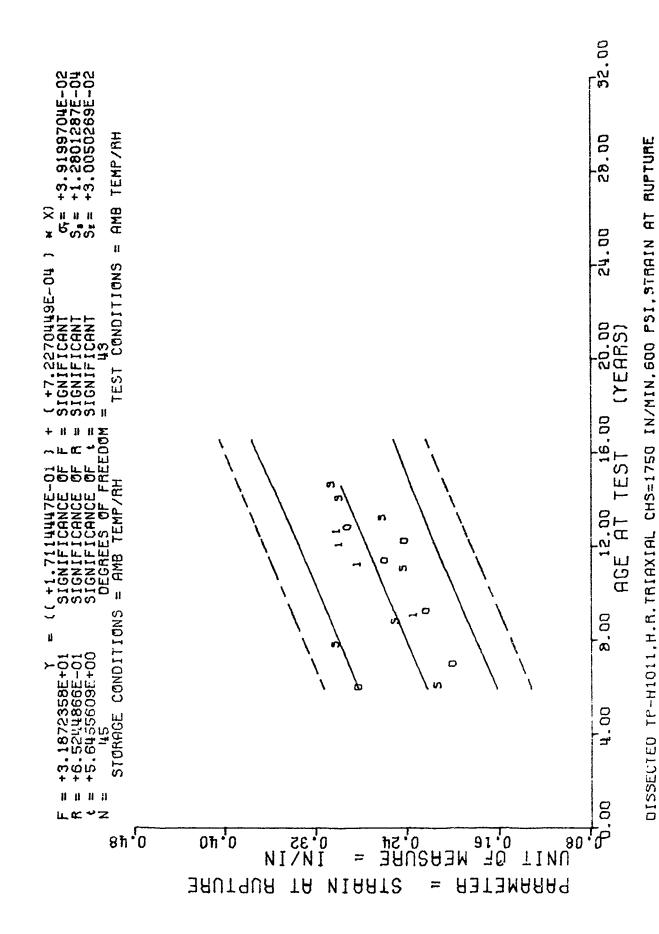
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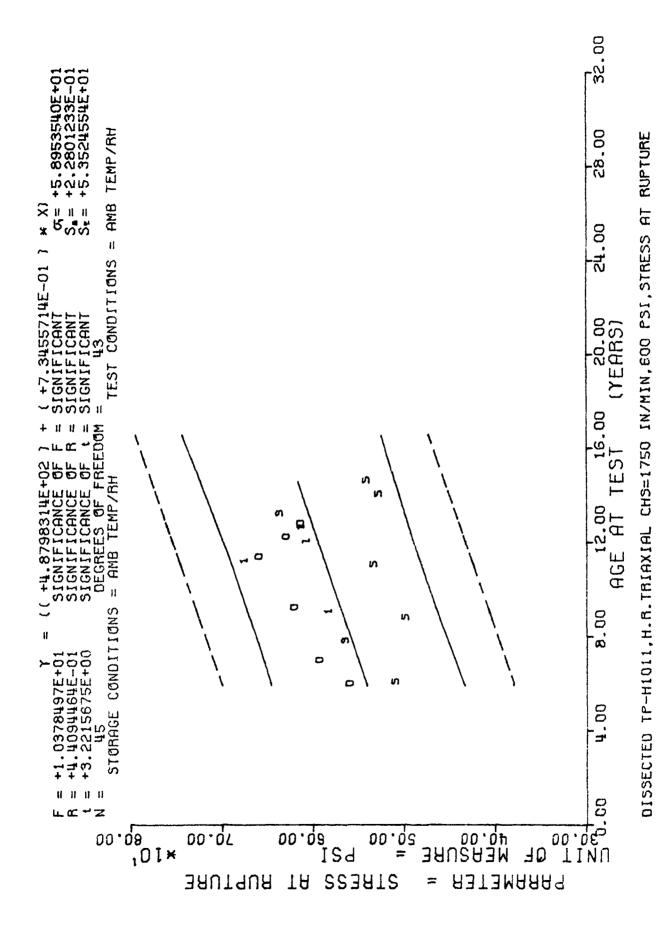
DISSECTED TP-HIDII, H.R. TRIAXIAL CHS=1750 IN/MIN, 600 PSI, STRAIN MAX STRESS

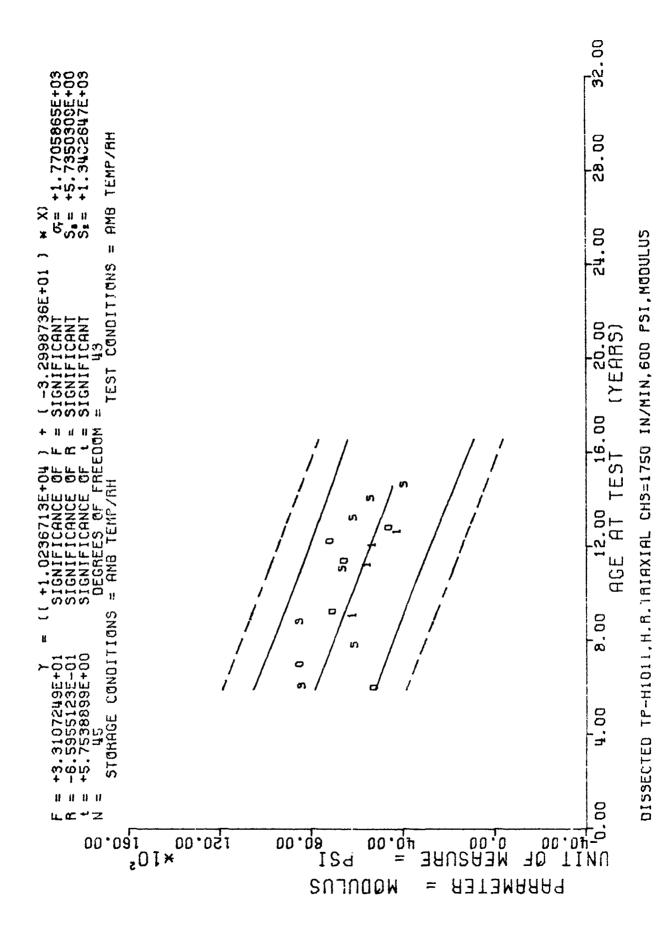
This sample size summary is applicable to figures 16 thru 20











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•	4	9.4.	13	16.	_	54.	56.	66.	<b>α</b>	9.0

CASEBOND TE ISILE, STAGE 1 DISSECTED, CHS 0.2, CSA 0.75

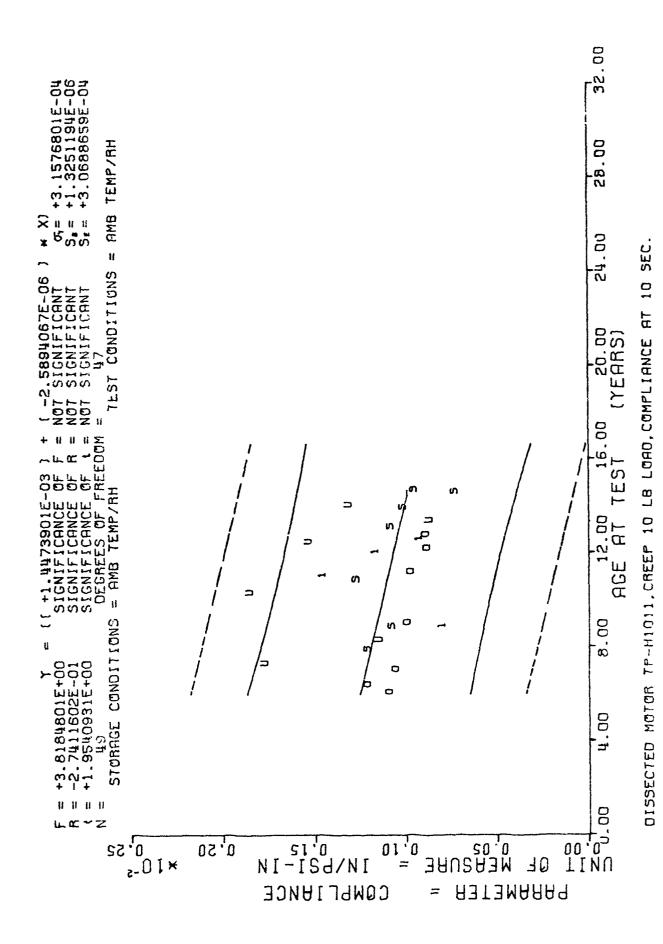
This sample size summary is applicable to figure 21

Figure 21

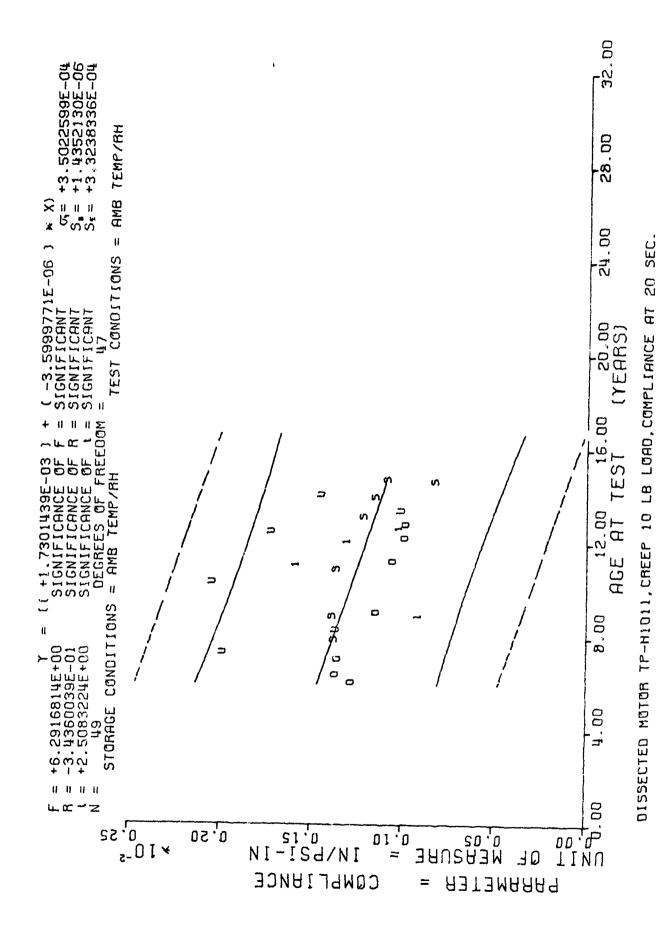
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AGE	(MONTHS)	8	•	4		•	6	•	8	30	•	5		•	•	6	-	3.			-	168.0	5	,

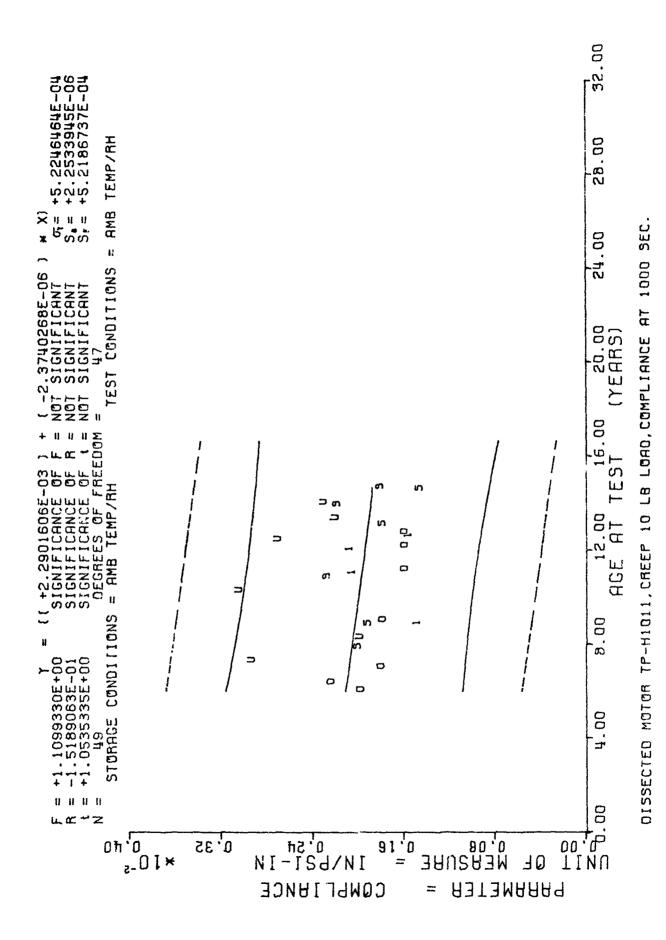
DISSECTED MOTOR TP-H1011, CREEP 10 LB LCAD, COMPLIANCE AT 10 SEC.

This sample size summary is applicable to figures 22 thru 24







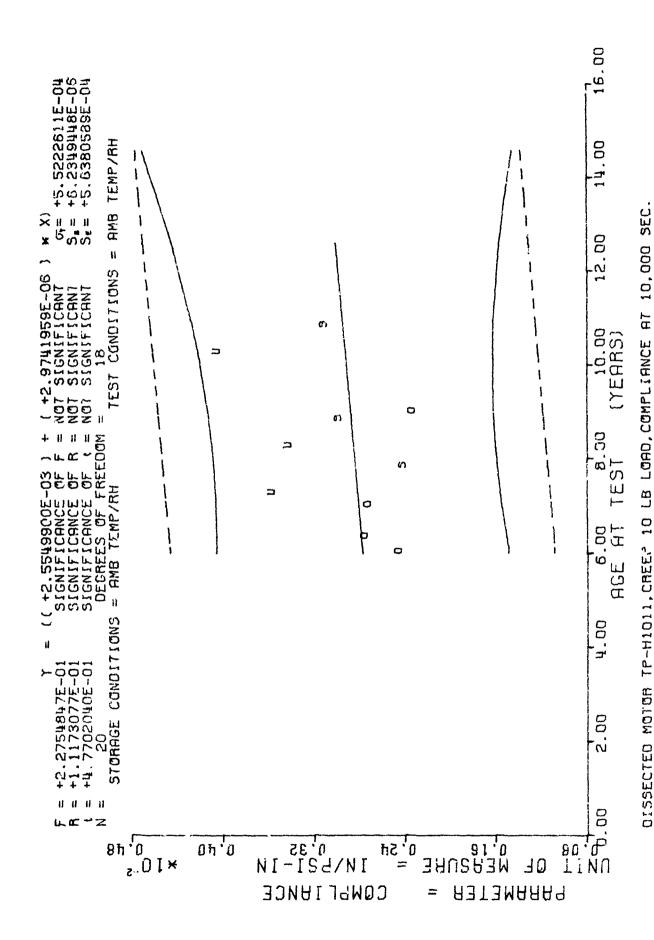


AGE NA IMONTHS) SAMPLES

SAMPLES	~	<b>m</b> )	~	~	~	ď	<b>F</b> )		<b>+</b> 1	~	•
(MONTHS)	8	76.0		87.0	4	6	106.0	8	123.0	ċ	C 141

DISSECTED MOTOR TP-H1011, CREEP 10 LB LOAD, COMPLIANCE AT 10,000 SEC.

This sample size summary is applicable to figure 25

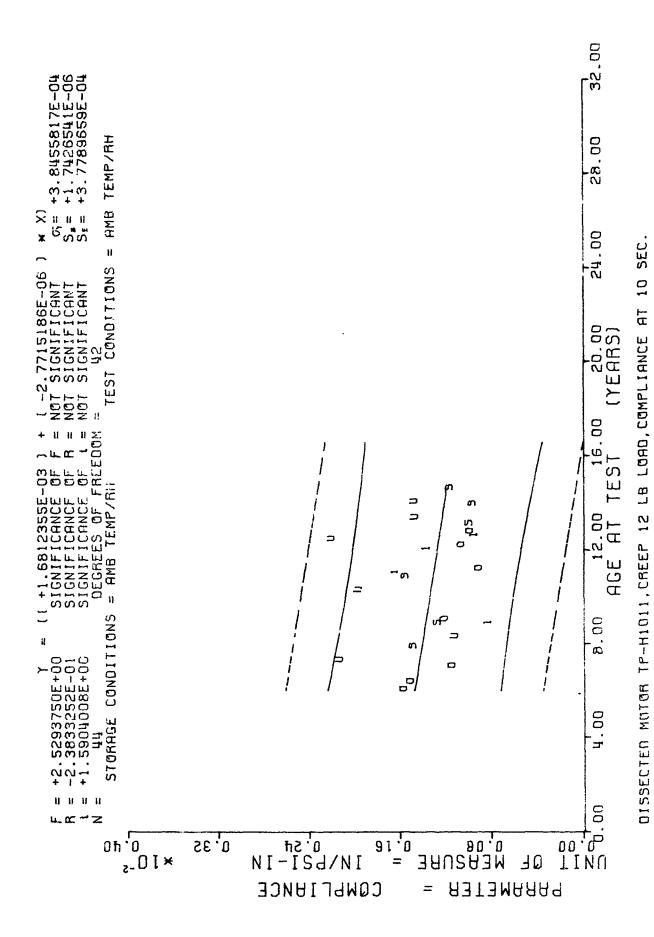


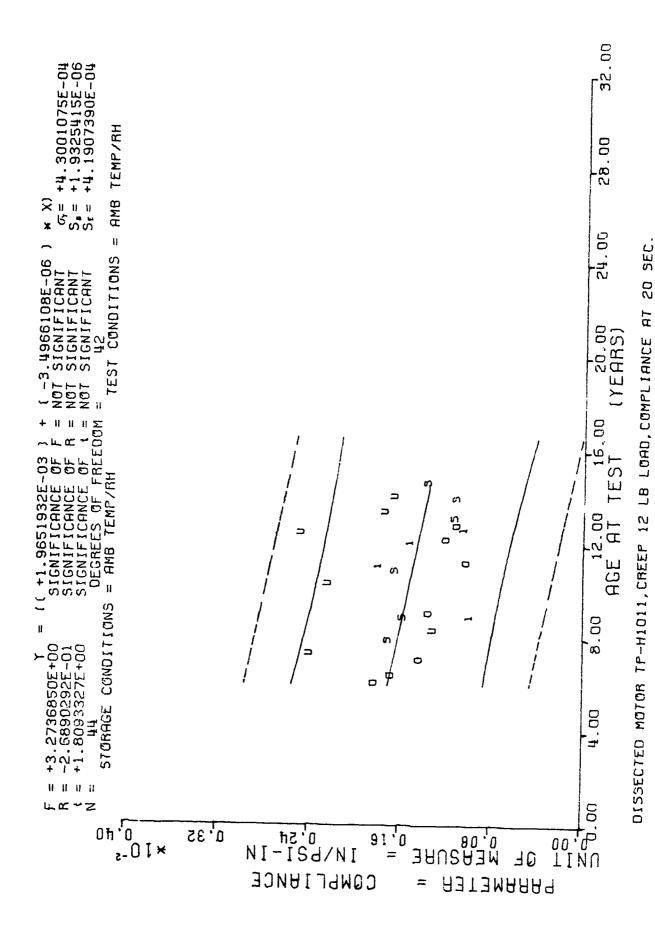
## \*\*\* SAMPLE SIZE SUMMARY ###

SAMPLES	<b>uruunum → → a</b>	
AGE (MONTHS)	72.0 76.0 84.0 87.0 94.0 99.0 106.0 133.0	4400 H W V O V B W

DISSECTED MOTOR TP-HI011, CREEP 12 LB LCAD, COMPLIANCE AT 10 SEC.

This sample size summary is applicable to figures 26 thru 28





DISSECTED MOTOR TP-HID11, CREEP 12 LB LORD, COMPLIANCE AT 1000 SEC.

Figure 28

COMPLIQUCE

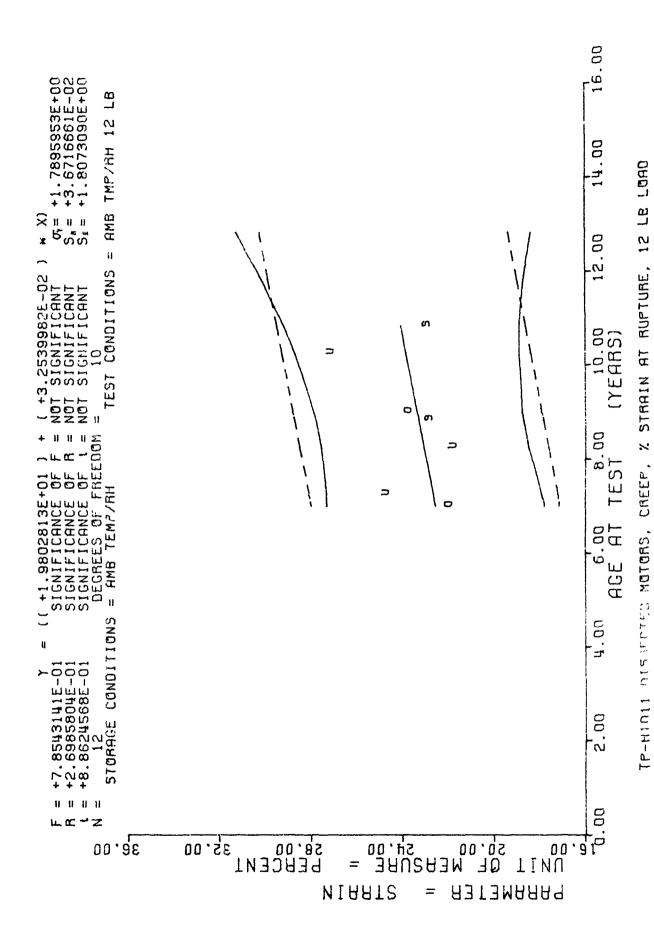
PARAMETER

## \*\*\* SAMPLE SIZE SUMMARY ###

Œ	SAVFLES	8	(N	~	<b>(")</b>	•	-	•
AGE	(MONTHS)	84.0	87.0	0.66	106.0	8	123.0	130.0

TP-HIGII DISSECTED MOTORS, CREEF, X STRAIN AT RUPTURE, 12 LB LCAD

This sample size summary is applicable to figure 29

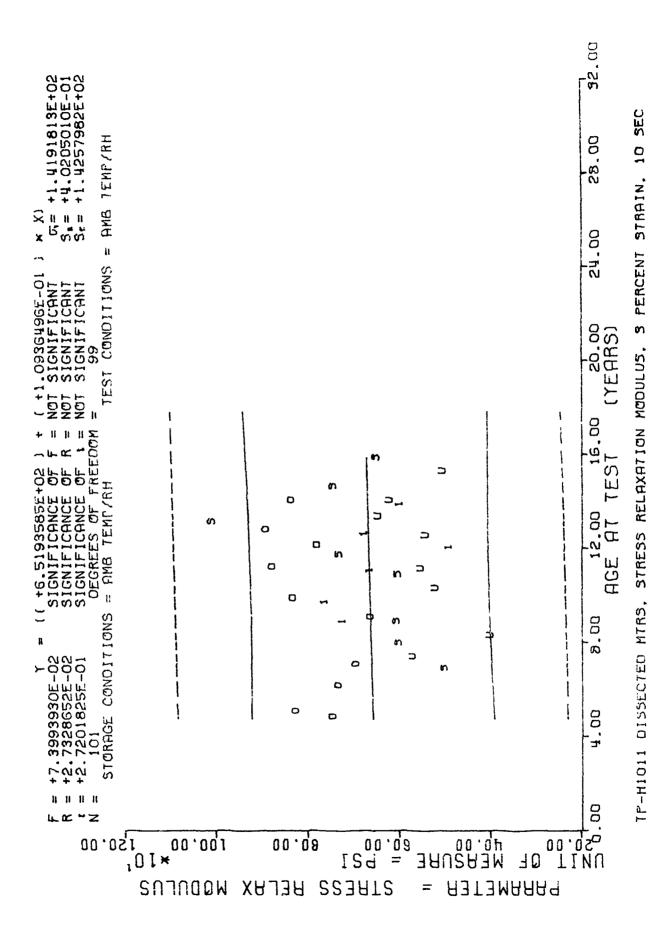


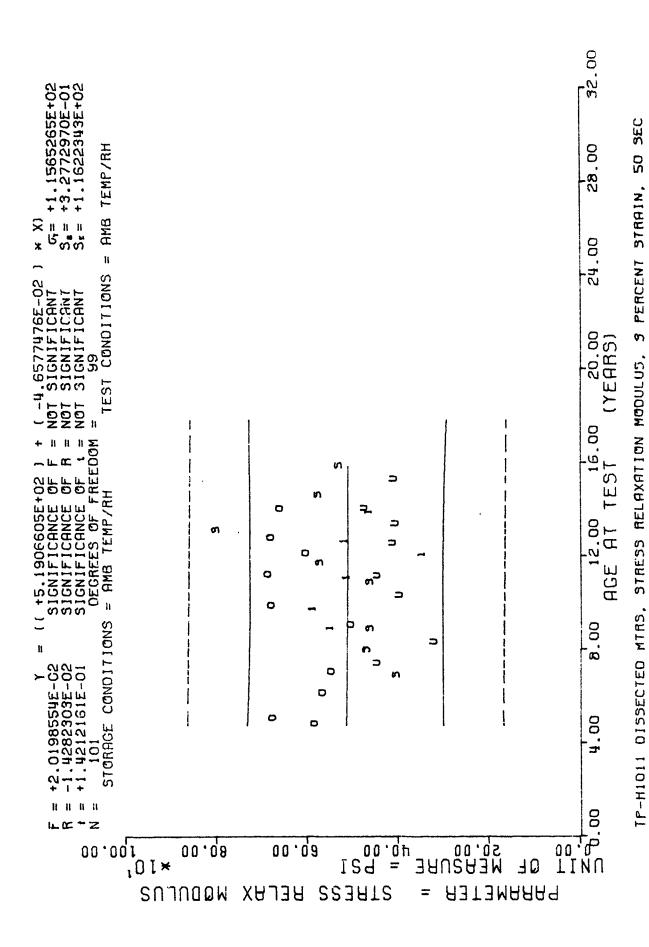
\*\*\* SAMPLE SIZE SUMMARY \*\*\*

2	SAMPLES	m	6	m	m	i ch	1																			
AGE	( MONTHS)	•	168.0	75.	œ	90.	• •																			
N.	SAMPLES	m	m	<i>(</i> 0,	m	m	m	4	m	ç	m	٣	٣	m	m	m	m	æ	~	4	m	m	m	~	ተስ	m
AGE	(SHINCH)	7.	0.09	3.	2.	84.0	88.(	95.0	6	106.0	108.0		118.0	123.0		132.0	133.0		140.0	144.0	Š	Ö	•		157.0	160.0

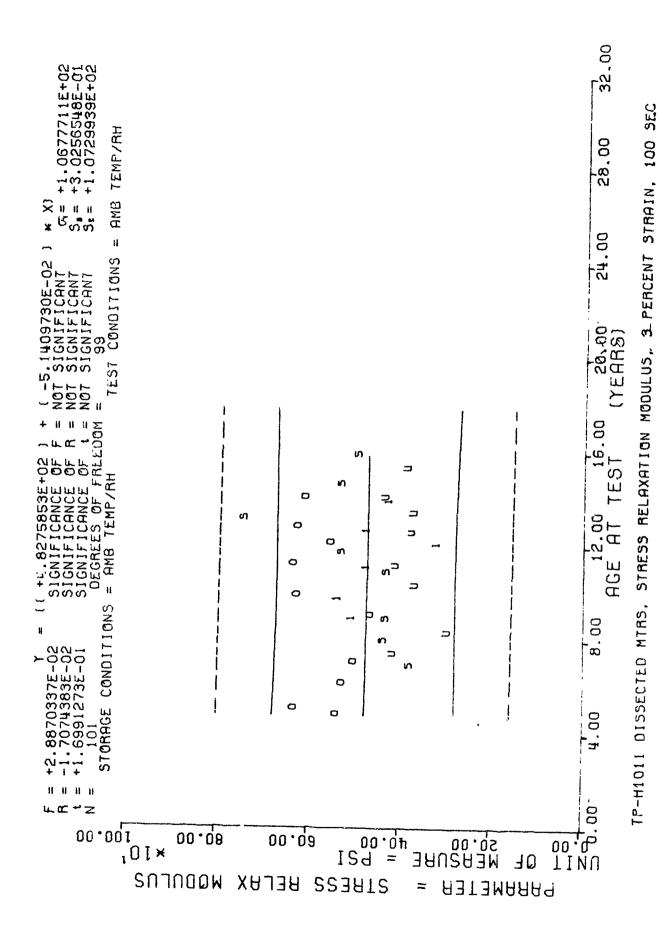
TP-HIGII DISSECTED MTRS. STRESS RELAXATION MODULUS. 3 PERCENT STRAIN, 10 SEC

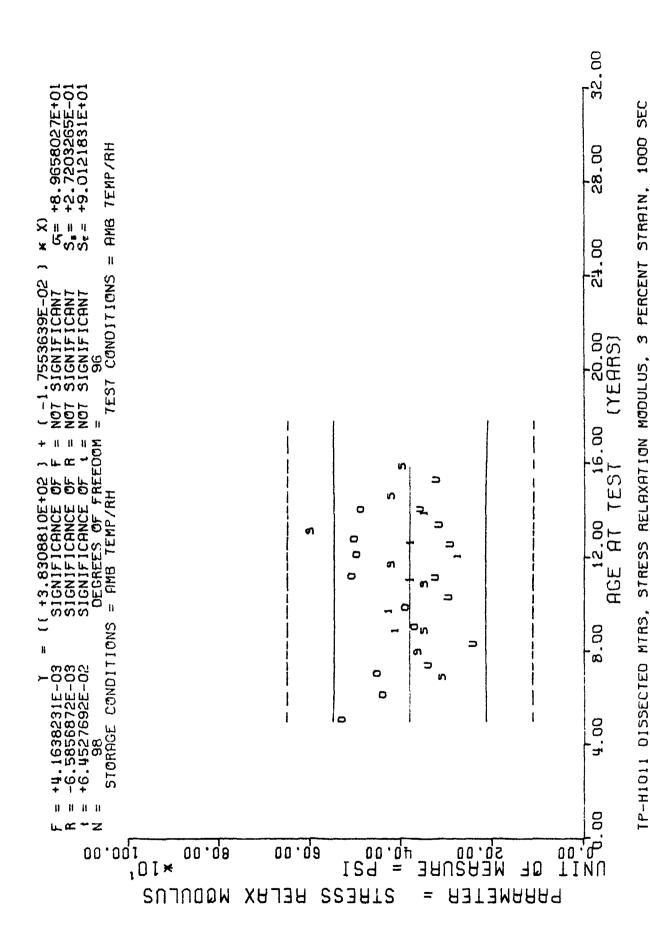
This sample size summary is applicable to figures 30 thru 33





- 54 -





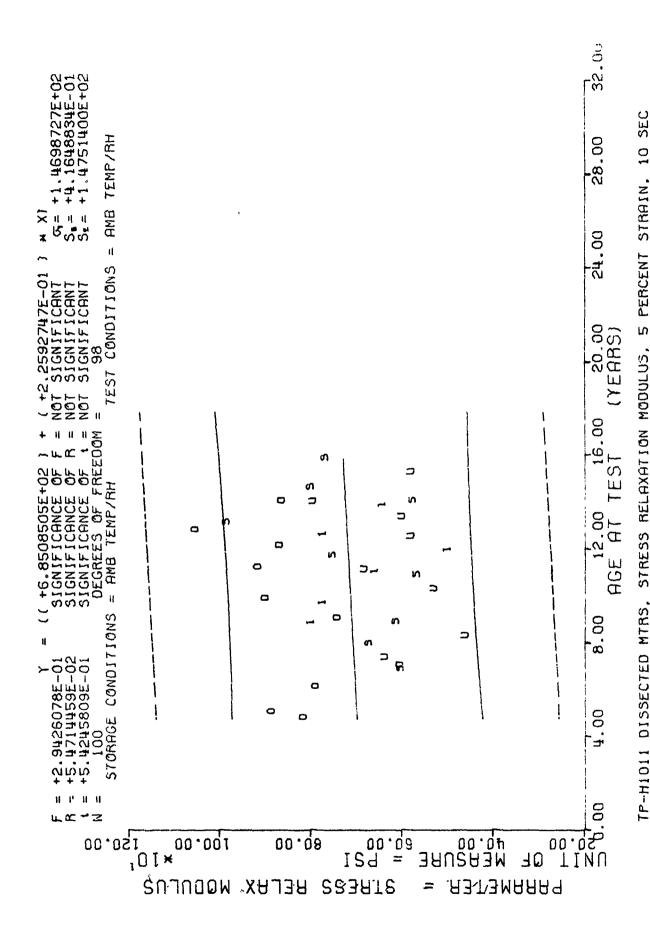
-- 56 **-**-

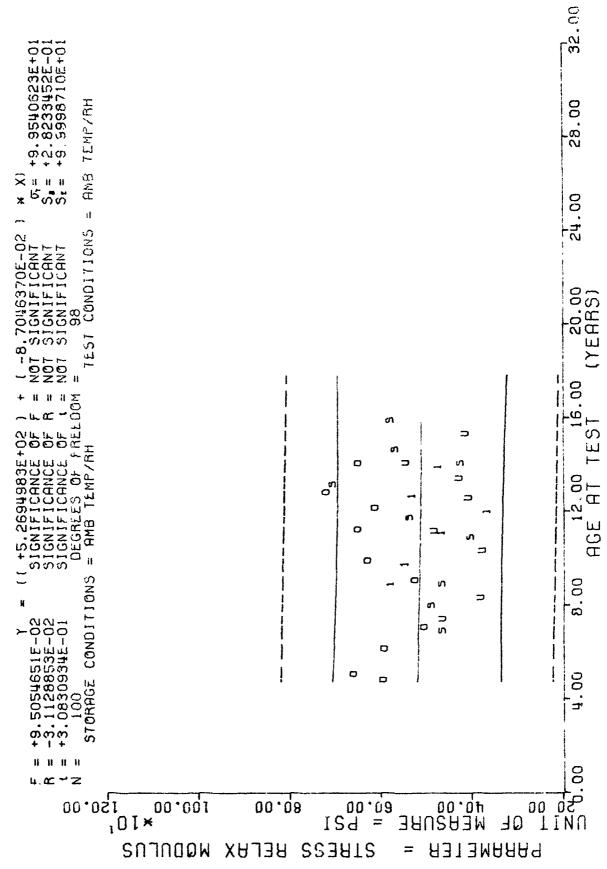
## \*\*\* SAMPLE SIZE SUMMARY \*\*\*

SAMPLES	<b>๛๛๛๛</b>
AGE (MONTHS)	166.0 168.0 175.0 193.0 190.0
SAMPLES	
AGE (MONTHS)	11 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1

TP-HIGHL DISSECTED MTRS, STRESS RELAXATION MODULUS, 5 PERCENT STRAIN, 10 SEC

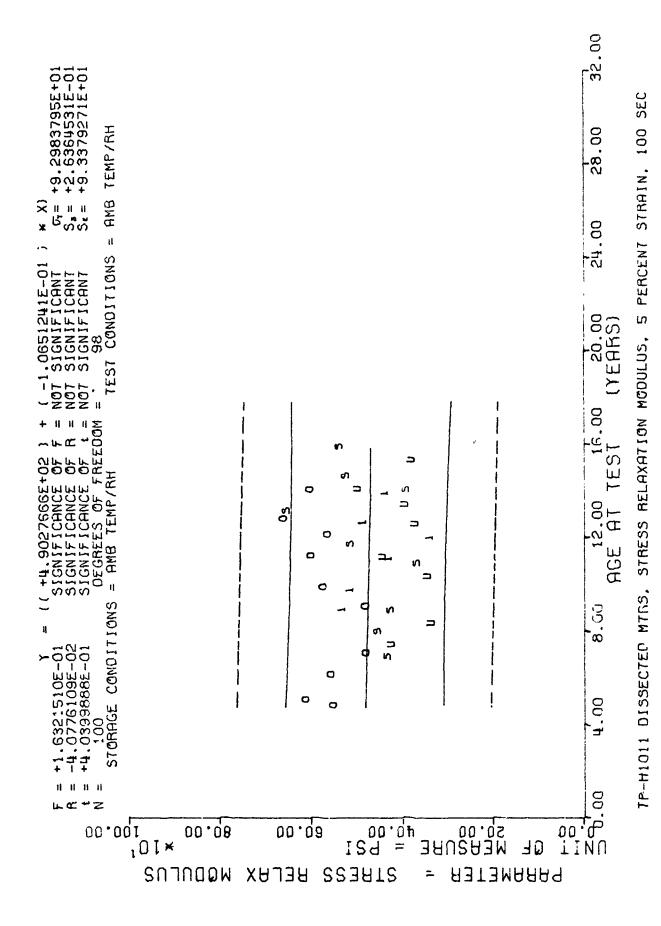
This sample size summary is applidable to figures 34 thru 37

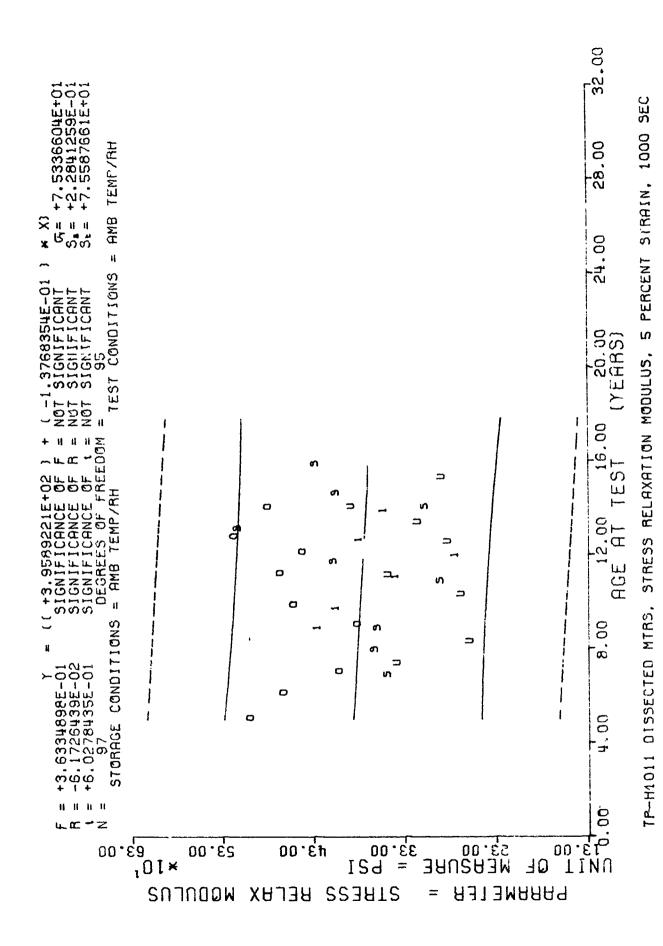




S PERCENT STRAIN, 50 TP-HIGH DISSECTED MTRS, STRESS RELAXATION MODULUS,

SEC

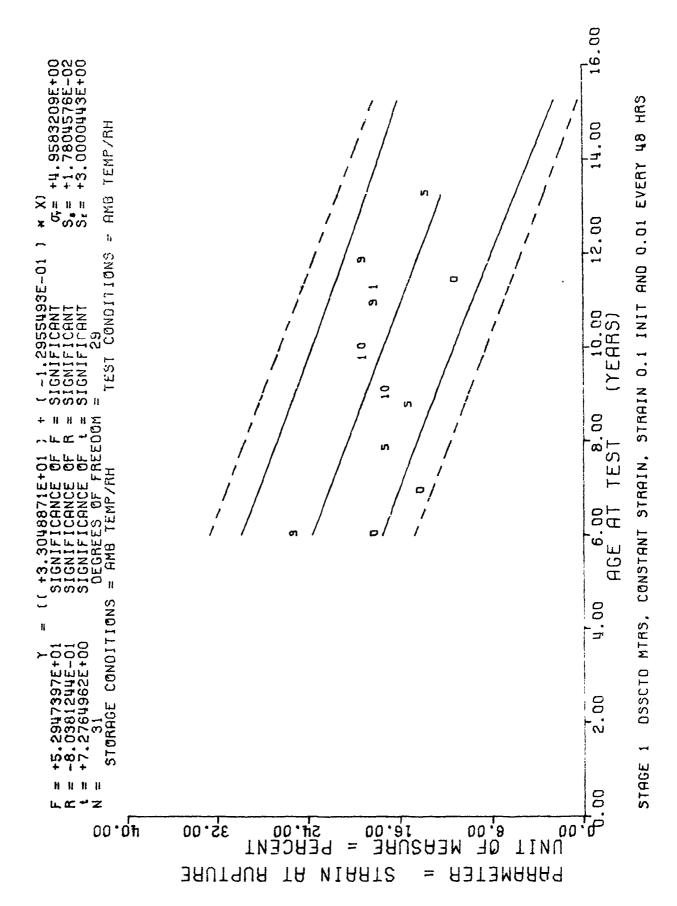




91		-	~	~	~	-	-	~4	2	2	<b>~</b> 4	2
~	3	4	9.50	57.	9.0	117.0	20.	31.	35.	37.	42.	Ġ

STAGE 1 DSSCIE MIRS, CONSTANT STRAIN, STRAIN 0.1 INIT AND 0.01 EVERY 48 HRS

This sample size summary is applicable to figure 38

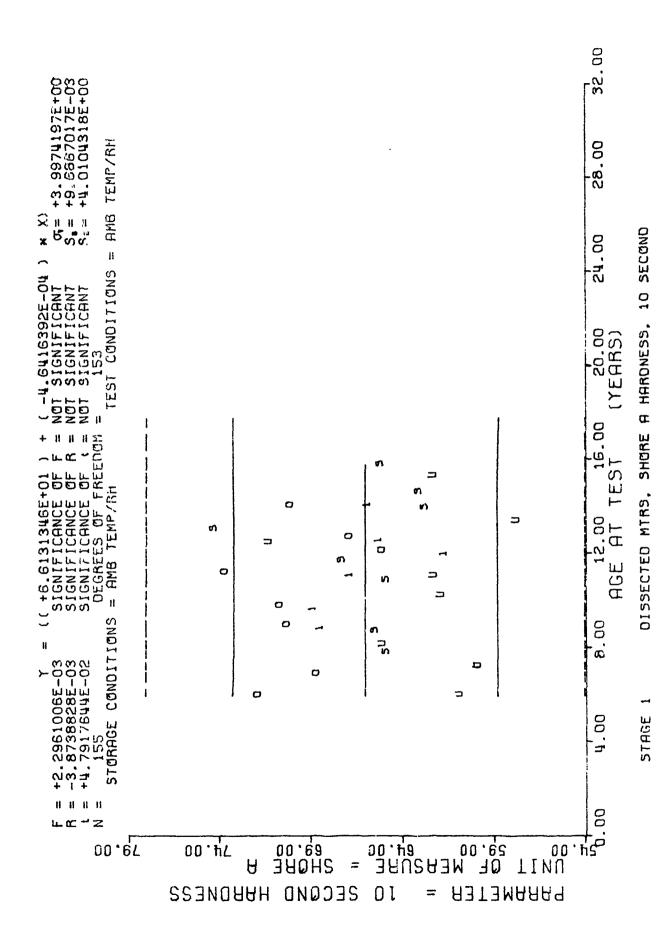


\*\*\* SAMPLE SIZE SUMMARY \*\*\*

NR SAMPLES	100
AGE (MONTHS)	183.0 189.0
MRSAMPLES	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~
AGE (MONTHS)	1100 1100 100 100 100 100 100 10

DISSECTED MTRS, SHORE A HARDNESS, 10 SECOND STASF 1

This sample size summary is applicable to figure 39



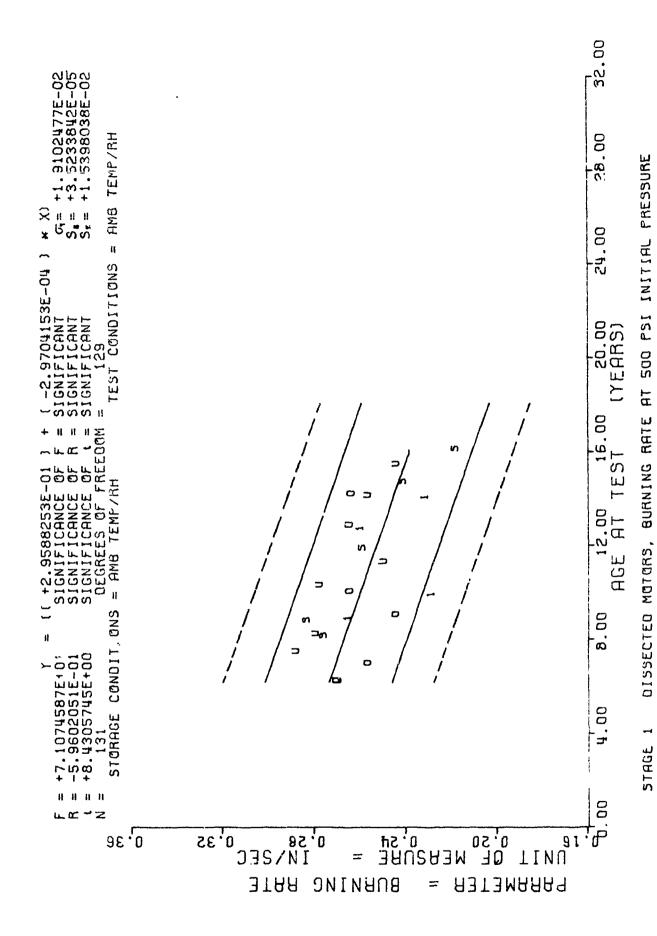
\*\*\* SA' .E SIZE SUMMARY \*\*\*

ď	SAMPLES
AGE	(SHINDW)

84 9 9 9	ນອວທະນອ	<u> </u>	000000 00000
4 40 4	98 95 96 13	200000000000000000000000000000000000000	168.0 169.0 170.0 135.0 195.0

DISSECTED MOTORS, BURNING RATE AT 500 PSI INITIAL PRESSURE STAGE 1

This sample size summary is applicable to figure 40

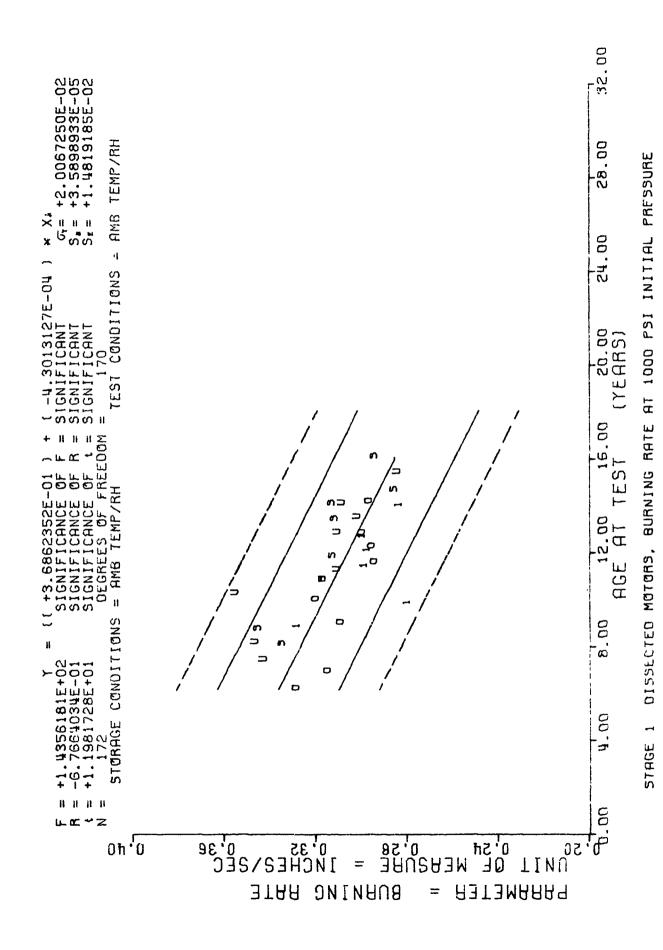


\*\*\* SAMPLE SIZE SUMMARY \*\*\*

NR SAMPLES		
AGE (MUNTHS)	176.0 185.0 193.0	
NR SAMPLES	~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~	כ
AGE (ND4THS)	11000000000000000000000000000000000000	•

STAGE 1 DISSECTED MOTORS, BURNING RATE AT 1909 PSI INITIAL PRESSURE

This sample size summary is applicable to figure 41

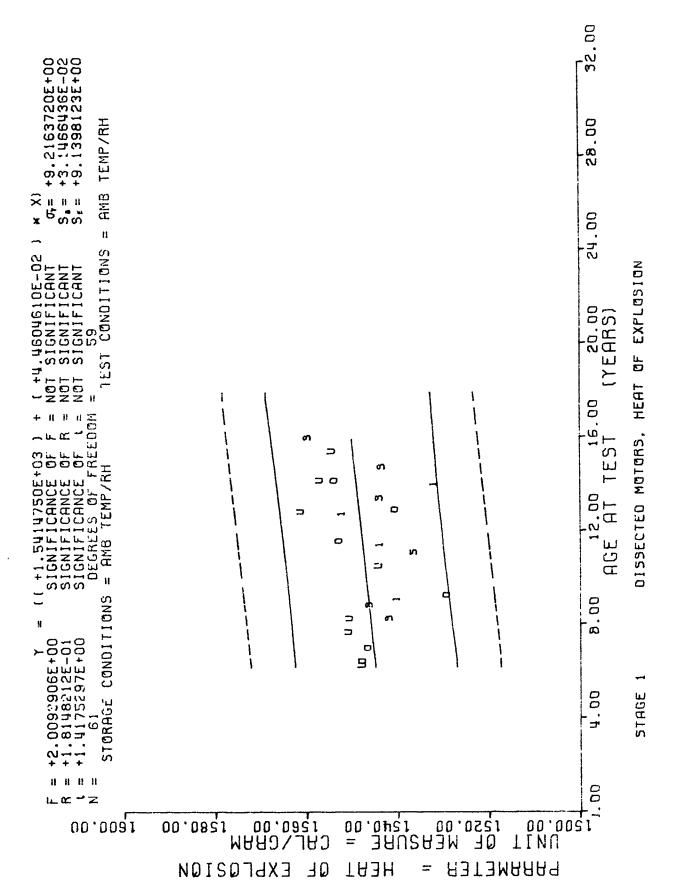


A.R.	SAMPLES
AGE	(M) 4 [HS]

m	m	m	6	ç	M			<b>-</b> 1		~	8	٣	٣	кт.	m	8	9	n	К	٤
4	Ş	3		$\infty$	)5.	38.	10.	25.	32.	36.	38.	152.0	53.	54.	5.7	57.	59.	76.	34.	<b>)</b> 1.

STAGE 1 OISSECTED MOTORS, HEAT OF EXPLOSION This sample size summary is applicable to figure 42

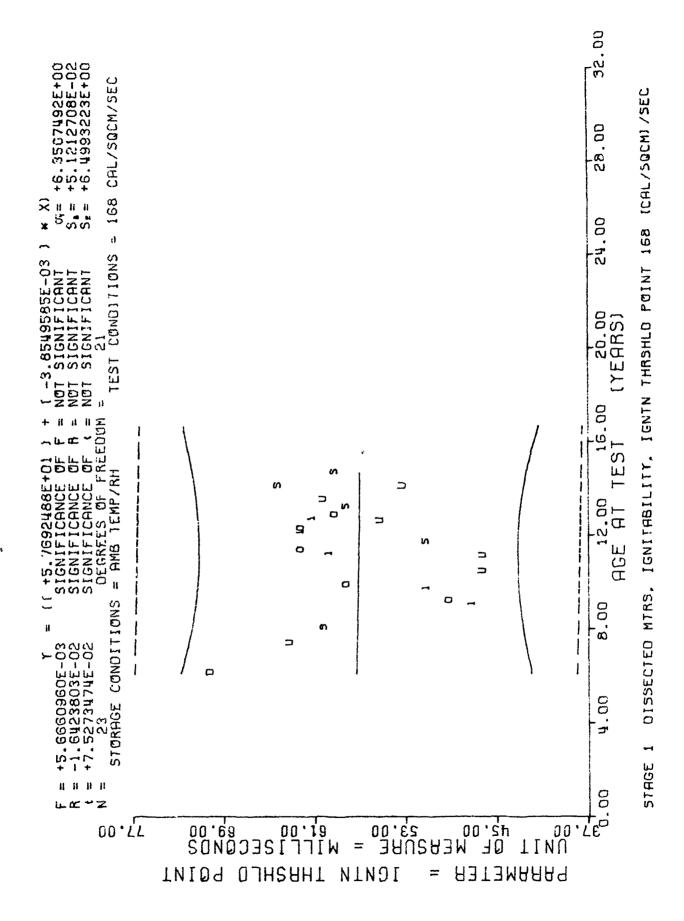
- 70 -



SAMPLES	~	<b>,-</b> 4	r <del>ed</del>		-	<b>~</b>	-		<b></b> 1	2	~	~4		-	<b>,1</b>	-4	<b>~</b>	7		2	rad
( MONTHS)	8	ဆ	9	9.8	10.	16.	2	25.	33.	34.	36.	4 0	45.	47.	51,	52.	54.	58	62.	159.0	76.

DISSECTED MTRS, IGNITABILITY, IGNTN THRSHLD POINT 168 (CAL/SQCM)/SEC STAGE 1

This sample size summary is applicable to figure 43

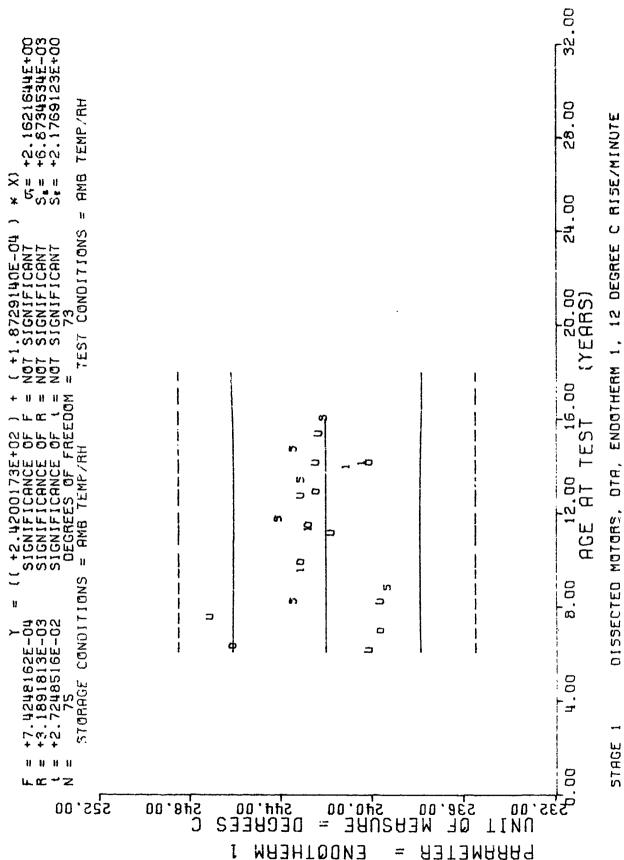


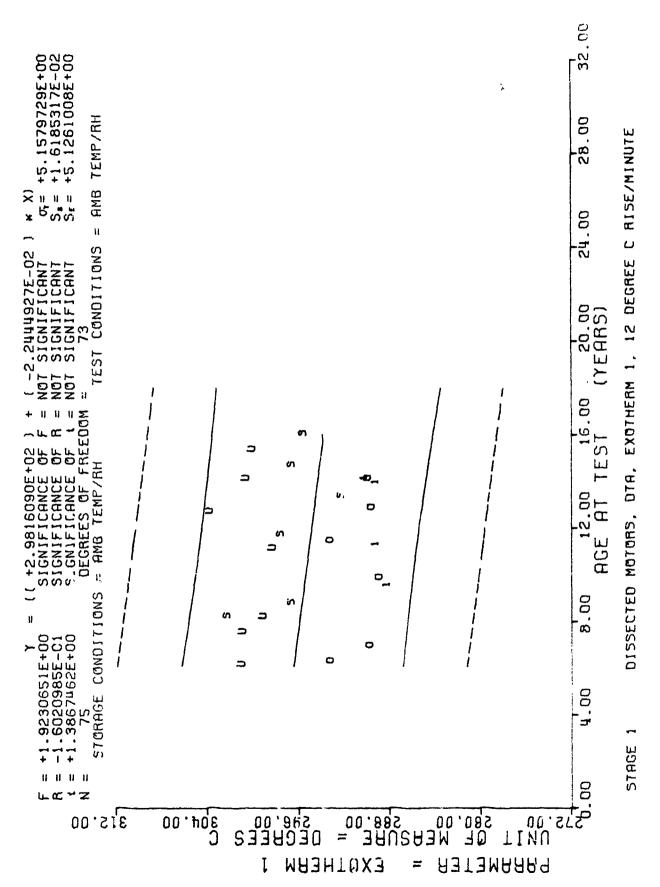
<b>4</b> M	<b>~</b> ~	<b>O</b> K	· m r	ો લ્લે ત	n m m	<b>~</b> ~	m m	& W R	n M
w 2	60	86	•	900	40,000	52.	60.	169.0	92.

DISSECTED MOTORS, DTA, ENDOTHERM 1, 12 DEGREE C RISE/MINUTE STAGE 1

This sample size summary is applicable to figure, 44 and 45

Figure 44





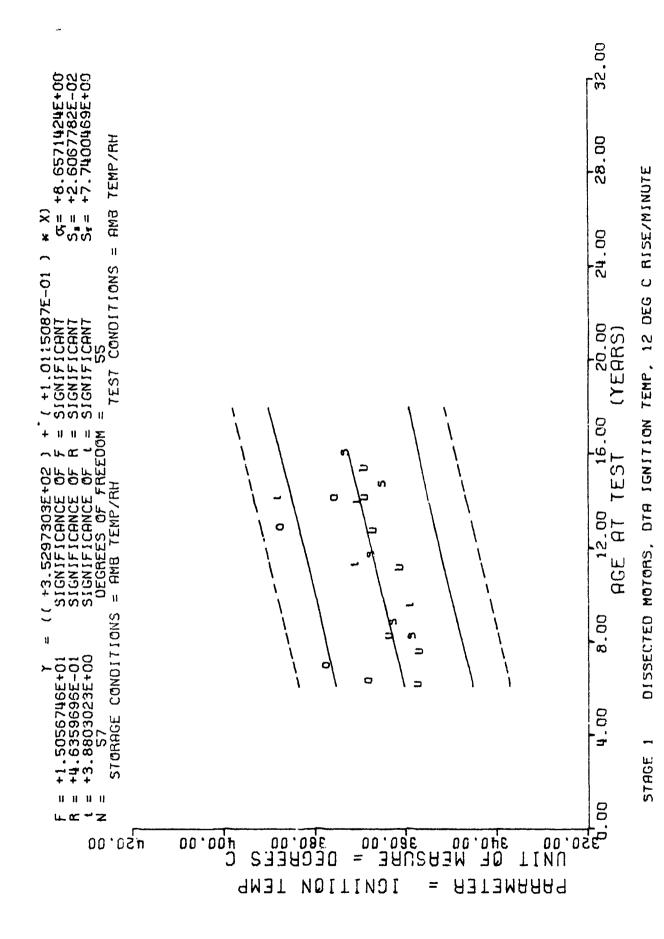
## \*\*\* SAMPLE SIZE SUMMARY \*\*\*

SAMPLES	4 4 4 4 5 5
AGE (MONTHS)	43.0 843.0 99.0 98.0

4	m	•	S.V.	~	m	~	m	N	~	N	40	~	¥n.	m
i	ö		'n	+	6	8	6	2	+	7	6	0.9	+	2
			0	-	(1)	3	4	8	50	•	•	17	8	0

DISSECTED MOTORS, DTA IGNITION TEMP, 12 DEG C RISE/MINUTE STAGE 1

This sample size summary is applicable to figure 46



## \*\*\* SAMPLE SIZE SUMMARY \*\*\*

Z	S) SAMPLES	•	4	*	*	*	*	4	•	•	•	•	•	*	9	•
AGE	NTH	33	35	46.	484	50.	152.0	54.	57.	63.	67.	69	70.	76.	84.	91.

## DISSECTED MTR, STAGE 1, TP-H1011, SOL GEL, PERCENT EXTRACTABLES

This sample size summary is applicable to figures 47 thru 50

Figure 47

Figure 48

SWELL RATIO

GEL

GEL.

Sal

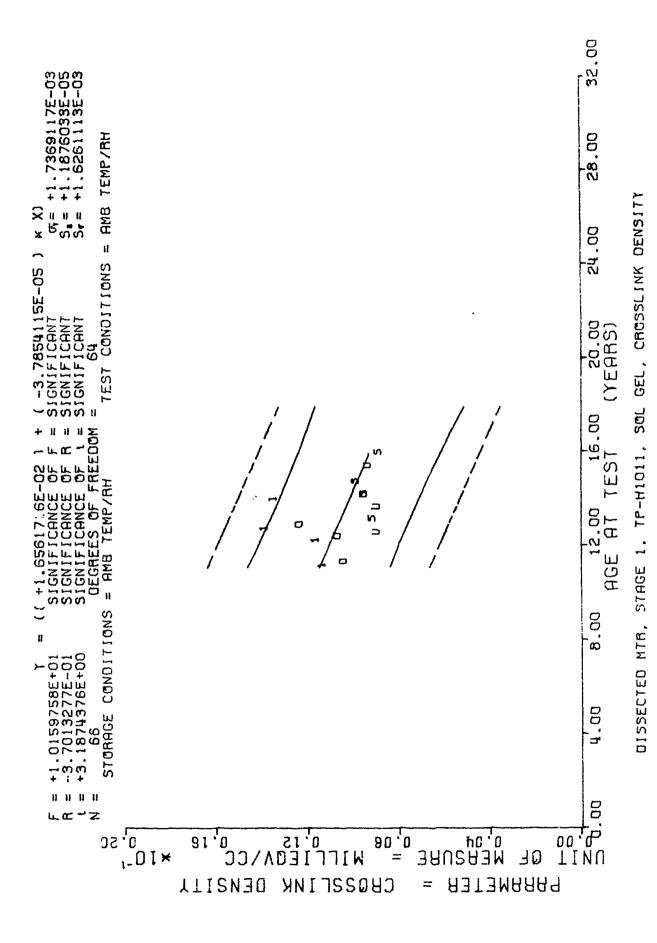
TP-H1011,

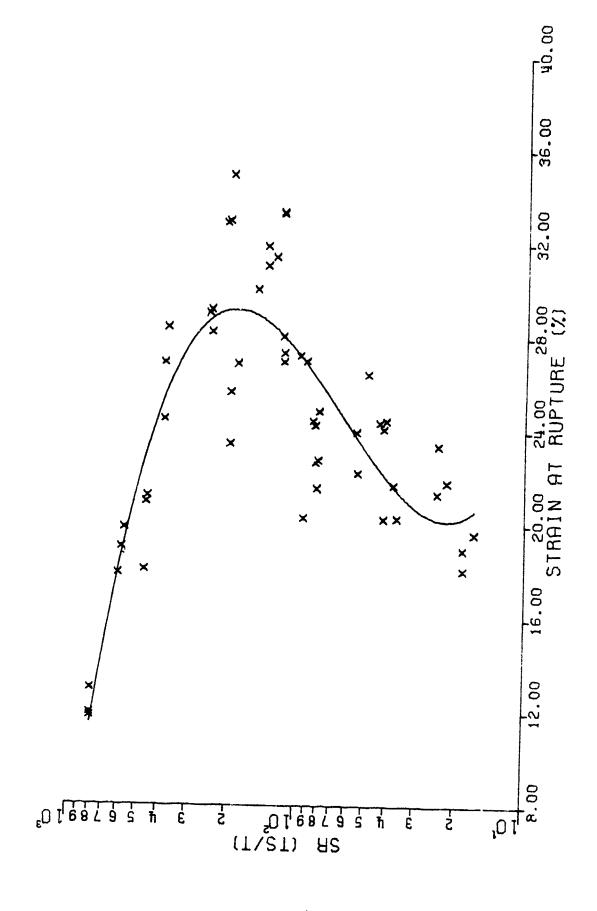
STAGE

MTR,

DISSECTED

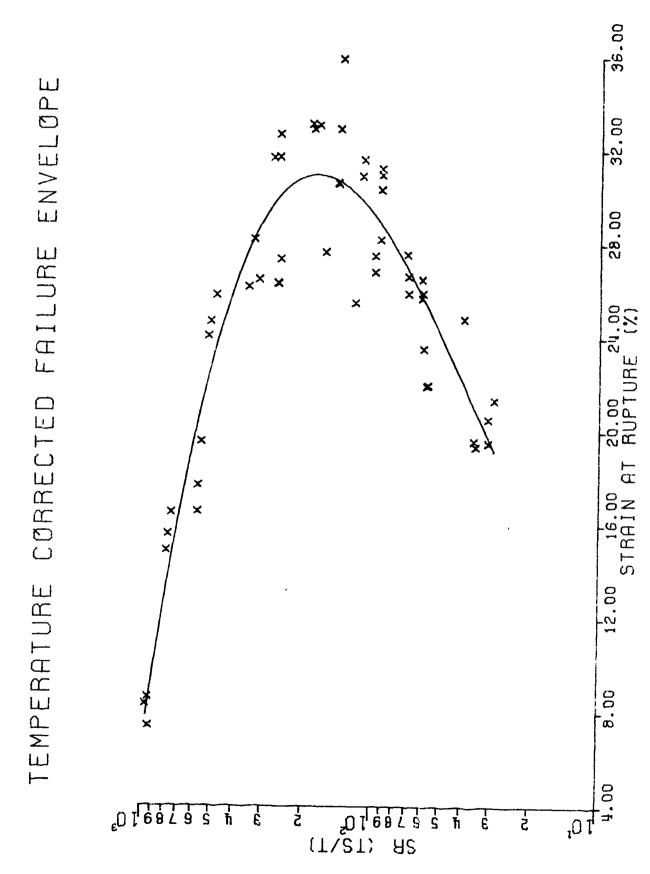
DISSECTED MTR, STAGE 1, TP-H1011, SOL GEL, DENSITY





TP-H1011 FAILURE ENVELOPE (MOTOR/SN STM-012) DISSECTED MOTOR, STAGE I.

Figure 52



FAILURE ENVELOPE (MOTOR/SN 0012199) DISSECTED MOTOR, STAGE I, TP-H1011

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	ž
Minuteman Solid Propellant	
)	
Testing was performed to determine the useful shelf/service life for LGM-30 Stage I Rocket Motors. A three year storage program for propellant and components was started in May 1961. This program was then extended to a ten year study and later continued indefinitely to assure that a deterioration in motor physical characteristics could be detected in time to take some corrective actions before the weapon system performance deteriorated below an acceptable level.  (over)	

DD 1 JAN 73 1473 EDITION OF 1 NOV 65 IS OBSOLETE - 87 - SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

This report covers only propellant data and limited case bond data. The malfunction of an environmental chamber destroyed component samples that had originally been part of this testing program (and the inadvertent burning of some motors during dissection reduced the material available for testing). Planned dissection of selected motors in the future will provide samples for continued component testing. Test specimens for this reporting period were obtained from motors STM-012 0012099, 0012199 and UP-7775 block propellant.

Separate analyses were made on the respective motors and block propellant for the first time in this report and are shown in the regressions. The plotting symbols for each motor and block propellant are listed in the statistical analyses section.

The data from this test period was combined with data from previous testing and entered into the GO85 computer for storage, analysis, and regression analysis. From the statistical analysis of all data tested to date, significant degradation of the propellant does not appear likely for at least two years past the oldest data point.

Future testing will be conducted on dissected motors.